

Deutsche Gesellschaft für Luft- und Raumfahrt Lilienthal-Oberth e.V.





Verein Deutscher Ingenieure Hamburger Bezirksverein e.V. Arbeitskreis Luft- und Raumfahrt

Invitation to an RAeS lecture in cooperation with the DGLR and VDI

#### **Aircraft Fire and Evacuation Simulation**

**Prof. Edwin Galea,** Director, Fire Safety Engineering Group, University of Greenwich

Lecture followed by discussion Entry free ! No registration required !

Download: http://hamburg.dglr.de



Date: Thursday, 16th October 2014, 18:00 Location: HAW Hamburg Berliner Tor 5, (Neubau), Hörsaal 01.12



**Exploring the Appropriateness of the Aviation Industry Evacuation Certification Requirements Using Fire and Evacuation Simulation** 

> Prof Ed Galea Fire Safety Engineering Group (FSEG) University of Greenwich







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#### FSEG: Modelling Safety and Security

- FSEG was Founded in 1986 by Prof Galea in response to the Manchester Airport B737 fire.
- Today it consists of 30 researchers including:
  - fire engineers, CFD specialists, psychologists, mathematicians and software engineers.
- Research interests include the mathematical modelling and experimental analysis of:
  - evacuation dynamics in complex spaces,
  - pedestrian dynamics in complex spaces,
  - combustion and fire/smoke spread,
  - fire suppression,
  - homeland security
- Application areas include:
  - aerospace, built environment, marine and rail.



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## **Applications of FSEG Software**



A380 – Super Jumbo



Millennium Dome



**Stadium Australia** 



**Rail Stations** 



**Royal Ascot** 

ittany Ferries

Large PAX Ships



**Naval Ships** 

**Pentagon Shield** 



**Historic Buildings** 

**Beijing Olympic Stadium** 



**Rhode Island** 





WTC 9/11 analysis

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# □ Post-crash Fires

- One of the most serious threats to passengers in survivable aircraft accidents.
- Initial external fuel fire spreads into aircraft interior either via ruptures or burn-through.
- For conventional aircraft, time to Flashover is a critical factor for evacuation and survival.

## **Certification Trial**

- **50%** of exits available, one from each exit pair.
- **90 seconds** maximum allowable time for evacuation

## Varied Openings in Accidents

- In real incidents, various number of exits are likely to be available for evacuation.
- Fire and evacuation modelling can be used to investigate the impact of accident events on survivability and can also be used for certification analysis.



Thailand Phuket, 16/09/07



Toronto, Canada 02/08/05



Japan Okinawa, 20/08/07



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### **FSEG Evacuation Data Collection**



Aircraft evacuation exit behaviour – Type A



Aircraft pax stair behaviour



Aircraft evacuation pax behaviour in smoke



Upper deck Type A slide behaviour







# BWB pax exit selection



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## airEXODUS Software

- Developed by FSEG and under constant development since 1989
- Agent based model with Rule Based Behaviour.
  - •Behaviour is adaptive
  - •Some rules stochastic.
- Behaviour model considers:
  - •People-people
  - •People-fire
  - •People-Structure
- airEXODUS unique features include:
  - ability to simulate impact of heat, smoke and toxic gases on evacuation capability of individuals
  - ability to include interaction of crew with paxs
  - extensive validation history



movement



## airEXODUS — TOXICITY MODEL

- **Toxic, Irritant and Physical Hazards include:** elevated temperature, thermal radiation, HCN, CO, CO<sub>2</sub>, low O<sub>2</sub>, HCL, HF, etc.
- Physiological impact of narcotic gases / temp / radiative heat determined using **FED Toxicity Models**. Impact of irritant gases determined through **FIC Model**.
- **FED Models** effects of narcotic gases related to *dose* received rather than exposure *concentration*.
- FIC Models effects of irritant gases related to conc
- **Incapacitation** determined to occur when the ratio of dose received (heat or toxic gases) over time to dose required to cause incapacitation reaches unity
- As occupant moves through **Smoke**, travel speed is reduced according to experimental data of Jin, representing impact of reduced visibility.
  - Occupants will crawl when smoke conc. or temp at head height exceed critical values.



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## airEXODUS Applications





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#### airEXODUS ations

## airEXODUS Applications

- aircraft design, including innovative concepts such as BWB
- demonstrating compliance with 90-second certification requirements,
- crew training,
- development of crew procedures,
- resolution of operational issues and
- accident investigation.
- Aviation applications include:
  - Airbus e.g. A380, A340-600, BWB
  - Bombardier e.g. Dash 8-400, RJ, concept aircraft
  - Mitsubishi e.g. MRJ
  - BA e.g. Novel cabin layout for B777
  - Jet Aviation e.g. VIP configurations
  - Zodiac e.g. novel business class cabin



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### Validation of Egress Simulations





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## **Blind Validation Results**

airEXODUS predicted exit usage reflected in the experimental results.



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## **SMARTFIRE Software**

- Developed by FSEG and under constant development since late 1980s
- Software is CFD based.
- Main features include:
  - Two equation K EPS turbulence model
  - 24 ray discrete transfer radiation model.
  - Eddy Dissipation combustion model.
  - Advanced flame spread model for aircraft fire simulation.
  - Advanced toxicity model for predicting generation of toxic species in fire.
  - Unstructured mesh capability for dealing with complex curved geometries
  - Parallel implementation capable of utilising standard PC computers connected via Ethernet.
  - Developing a GPU version with speedups of 30X achieved.





#### SMARTFIRE Aircraft Fire Validation Model Application — simulation of C133 FAA test

- > Time to flashover
  - Often taken as the time when upper layer temp exceeds 600°C
  - In this study: time when the seat (Row 5) top temperature rapidly escalates
  - Experiment: **210** seconds
  - Prediction: 225 seconds
- > Flame front













#### At 20 seconds

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RAeS Hamburg Branch 16 Oct 2014 At 180 seconds

#### buildingEXODUS and SMARTFIRE Simulation of Station Nightclub Fire

Link fire simulation directly with evacuation analysis
Directly expose agents to developing hazard environment
Predict fatalities and injury levels.



- Last survivor evacuates after approx 127 seconds.
- Simulation predicts :
  - •84 fatalities compared with 100 in actual incident.
  - •25 serious injuries, of which 6 are life threatening.



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- Fire Simulation Results Scenario 3 Light Extinction Coefficient (2.30/m is equivalent of visibility of 1 m)
  - Over first 100 sec, poor visibility restricted to cabin section containing rupture
- Temp at 1.7 m above floor (skin burns occur at temp above 120°C, incapacitation follows after 1 minute exposure to 190°C).
  - Over first 100 sec, dangerous temp restricted to cabin section containing rupture





#### **RED: 185<sup>o</sup>C:** GREEN: 100<sup>o</sup>C; LIGHT BLUE: 58°C



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Average egress time for 33 69.3 sec, a slight increase over 30 01 65.9 sec.
 12 fatalities and 25 pays injured due to heat exposure. 3 of these have life.

 12 fatalities and 25 paxs injured due to heat exposure, 3 of these have life threatening injuries (FIH>0.6)



#### **EVACUATION CERTIFICATION** International Evacuation Certification Trial requires:

- 50% of exits available, one from each exit pair.
  - It is assumed that post-crash external fire occurs on one side of the aircraft and so it is further assumed that all the exits on that side of the aircraft are unavailable.
- 90 seconds maximum allowable time for evacuation.
  - It is assumed that after 90 seconds, conditions inside the cabin are non-survival or that flashover has occurred.
- Applied to all passenger aircraft.
  - Size or configuration of aircraft irrelevant, so same rules apply to A320 and A380.







Manchester, UK



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FAA test 1989

22/08/85

#### **PROBLEMS WITH EVACUATION CERTIFICATION**

## ☐ (Some) Shortcomings of Certification Trial

- Certification trial exit configuration is **not representative**:
  - most survivable accidents involve a different exit combination from that used in the certification trial [Galea 2006]
- Certification trial exit configuration is **not challenging:** 
  - with 50% exits available for evacuation, the standard certification trial exit configuration results in the shortest evacuation time [Galea 2007]
- 90 second requirement is **arbitrary** 
  - Influence of fire on survivability is not considered
- Certification trial is **not robust**:
  - Only a single evacuation trial is conducted which cannot provide a robust representation of the natural variation in the evacuation process.



## **AIM OF THIS STUDY**

□ For an aircraft configuration that has satisfied the FAR25.803 certification requirements, investigate the deficiencies of the certification trial as a safety indicator in post-crash situations involving fire

## **Cabin Configuration**

- Narrow body aircraft similar to B373 or A320
- Three exit pairs (T-C/B, T-III, T-C/B)
- Seating for 149 passengers

#### **Gamma** Fuselage Rupture

- Assume a cabin rupture located between the L2 and L3 exits
- Size of rupture is 0.89 m wide and 1.65 m high (the area of a T-B exit).





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## **TWO EXIT SCENARIOS**

#### **Two Exit Scenarios** are considered:

Scenario	S1	S2
Open exits	R1, R2 and R3	R1, R2 and L1

- S1: normal certification exit configuration
- S2: exit combinations commonly found in real accidents, e.g. Manchester Airport B737 fire, 1985.



Cabin configuration



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## FIRE SIMULATION SET UP

## **Geometry Set Up**

- External fire volume is 2.5m by 3.0m; HRR 7.8 MW, so that flame temp are close to experimental values of 1,480 K
- Mesh: 149,496 computational cells
- Red: external fire volume; Green: seats; Brown: walls; overhead bins removed for good visualization

## **Fire Models**

- Flame spread model for ignition of solid surfaces;
- Eddy Dissipation Combustion model for release of heat due to combustion of gas fuel generated by pyrolysis of solid materials;
- **48-ray Radiation model** for exchange of heat due to radiation;
- **Toxicity model** for the generation and transportation of toxic fire gases;

External fire



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SMARTFIRE set up

## **SIMULATION SET UP**

#### **Material Properties**

- Same as those in the previous work of C133 cabin fire simulations [Wang et al, 2013]
- **Molecular structure:** Epoxy  $CH_{1.3}O_{0.2}$  for all materials; Eexternal Kerosene fire is represented as Epoxy
- Heat release rate

A hypothetical heat release rate curve, derived from small-scale experimental tests; satisfies the criterion of  $65 \text{ kW/m}^2/65 \text{ kW}$ -

min/m<sup>2</sup>



• Other model parameters such as density, conductivity, specific heat, ignition temperatures, yields of toxic gases, etc. derived from various publications.



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## **IMPACT OF OPEN EXITS ON FLASHOVER**

#### **HRRs and Times to Flashover**

- Similar HRRs up to 250 seconds for the two scenarios;
- Onset of flashover: defined as the time at which the predicted interior HRRs rise sharply
- Time to flashover >> 90 seconds certification requirement



Scenarios		Time to flashover (s)	
<b>S1</b>		325	
<b>S2</b>		275	

Predicted HRRs from combustion of cabin interior material

S2 is more challenging than S1 (flashover occurs 18% sooner in S2) and both at least 3x longer than 90 s



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## **IMPACT OF OPEN EXITS ON HAZARDS**

## **Temperature (as Example)**

- Zone 17 in the aisle close to the over-wing exit
- With an air temp of 185°C, there is a 1 min survivability time
- Lower layer temp reach 185 °C: soon after onset of flashover in each scenario



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## **IMPACT OF OPEN EXITS ON HAZARDS** Temperature (as Example)



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#### **EVACUATION SIMULATIONS** Evacuation Set Up

- Type B/C exits in the front/rear and Type III over the wing
- **149 passengers** with response times no more than **8 seconds**;
- Passengers move to **their nearest viable exits**
- Results for each scenario represents average of **1000 repeat** simulations
  - 10 different certification compliant populations
  - Simulation repeated 100 times for each population
  - Passenger seating allocation is **randomised** in each simulation
- Egress times refer to **on-ground** times



#### Evacuation without fire

certification exit configuration **S1** 

realistic exit configuration S2



- Exit locations impact average travel distance: S1: 6.1m; S2: 10.2m
- Exit locations impact achieved flow rates, **58.8 ppm** for exit R1 in **S1** and **38.4 ppm** for exit R1 in **S2** 
  - the aisle is unable to supply sufficient passengers to keep both exits (R1, L1) working at full capacity in S2
  - Lower flow rate implies longer wait time in the ailse
- For Scenario S1, cabin can be emptied within 71.2s without fire; satisfying '90s requirement'; however, exit configuration Scenario S2 requires evacuation time of 98.1s, longer than '90s requirement';

#### S2 is more challenging than S1



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#### **Evacuation with fire**

Sc	enarios	Flow rate at R1 (person per minute)	Travel Distance (m)	Evacuation time (s)
<b>S1</b>		29.9	8.2	149.2
S2		13.5	12.3	260.8

- As noted in the evacuations without fire, S2 has lower exit flow rate and longer average travel distance than those in S1;
- Evacuation times in the presence of fire have greatly increased compared to the non-fire cases:
  - 149.2 s for S1,
    - S1 with fire is 2X as long as case without fire (71.2 s)
  - **260.8** for S2
    - S2 with fire is 3X as long as case without fire (98.1);

#### S2 is more challenging than S1



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Starting location (open symbols) and death locations (grey symbols) for a single S2 simulation

#### 1.2 fatalities in S1

- Located in seats adjacent to the rupture, died of heat from external fire, with a short survival time/travel distance;
- **14.6 fatalities in S2** 
  - Located in the rear of the cabin, died on the seats near the rupture, aisle, and places near the exits; with longer survival times/travel distances

#### S2 is more challenging than S1



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#### S2 is more challenging than S1



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#### **Wide-Body Analysis**

#### > Geometry

- > Aircraft is based on a A350 geometry as used in the AircraftFire FP7 project.
- Four pairs of Type A exits
- Composite fuselage and composite interior panels materials not necessarily those used in the A350.

#### Population

• 297 passengers and crew are generated from the default certification parameter set.



#### **Evacuation without Fires**

- The configuration satisfies the 90 s certification trial requirement in Scenario A (R1,R2,R3,R4) with an evacuation time 72.1 s
- Scenario C (L1, R1, L2, R2) produces an evacuation time of 140.4 s, much longer than 90s; the exit locations affect the travel distance and flow rate, subsequently the evacuation time







Local flashover occurs at 55 s, much sooner than the required evacuation time;
Not all the open exits are viable for evacuation

- Exit R4 is never used as it is engulfed by the external fire;
- Exit R3 becomes unviable after the cabin interior fire has developed;

>55 fatalities occur from 111.8s to 270.1s into the evacuation.





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## **Burn-through Scenarios with Wind**

- Evacuation with post-crash fire and wind
- Exits R1, R2, R3 and R4
- External fire makes R4 nonviable almost immediately.
- Aluminium fuselage: burnthrough occurs with flashover at 119 s and exit R3 becomes unviable; evacuation completed at 226s with 3.2 fatalities;
- Composite fuselage:
   evacuation completed after
   85s without any injuries.

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O() Spd: CLOCK : : : 1 2 3 4 5 6 7 8 9 10 000 **VIEXODUS** Composite http://fseg.gre.ac.uk

Aluminium

**Composite Fuselage, Fire and Wind – NO Burn-Through** 





Low CO concentrations at 1.7m above cabin floor at 600 s

- Exit R4 is engulfed
   Based on AircraftFire data, the composite fuselage does not burn through however, consider the impact of fire hazards on paxs during evacuation.
- Fire hazards heat and toxic gases (excluding SO<sub>2</sub>) from the investigated composite fuselage have negligible impact on evacuation
  - No passengers are injured by heat or toxic gases from the combustion of the composite fuselage;
  - The unavailability of Exit R4 (even it is opened in exit scenario A (R1, R2, R3, R4) affects the evacuation time
- > Irritants from composite materials
  - If SO<sub>2</sub> was produced in the same quantities as CO (as observed in FAA tests (Marker 2011)), it could have a significant impact on passengers during evacuation (approx 18 paxs are at high risk of incapacitation)
  - **Recommendation:** composite materials with a low yield of highly irritant compounds such as  $SO_2$  should be used in aircraft construction



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## CONCLUSIONS

- Evacuation performance has been compared between two different exit scenarios for an aircraft satisfying 90 second certification test;
  - A narrow body aircraft configuration similar to a B737 or A320.

S2:

- Study involved a given fire size and assumed post-crash fuselage rupture.
- The certification trial exit configuration **S1** is less challenging than the exit scenario **S2** which is more likely to occur in real post-crash accidents by producing:
  - Longer time to flashover (325 s > 275 s);
  - Shorter required evacuation time without fire (71.2 s < 98.1 s)
  - Shorter evacuation time with fire (149.2 s < 260.8 s); and
  - Fewer fatalities (1.2 < 14.6)



S1:

#### **CONCLUSIONS** The aviation industry certification trial requirement

- 50% of exits available, one from each exit pair.
- 90 seconds maximum allowable time for evacuation.

is inappropriate as a safety indicator as it is not:

- Representative of likely survivable accident exit configurations
- A sufficiently challenging exit configuration.
- The alternative exit combination which utilises the same number and type of exits as the current standard, but located in a more realistic accident combination provides a more Representative and Challenging exit combination and so should be used for certification applications.
  - I It can be argued that modelling should also be used for certification analysis and used to investigate additional exit configurations and additional repeat cases.



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## CONCLUSIONS

- Coupled fire and evacuation simulations can provide more insight into the fire safety of aircraft cabins than what can be found in certification trials
- Also provides significant insight when used forensically for accident analysis.

