



Aircraft

Landing Gear

The Evolution of a System

Gerd Roloff
Systems General

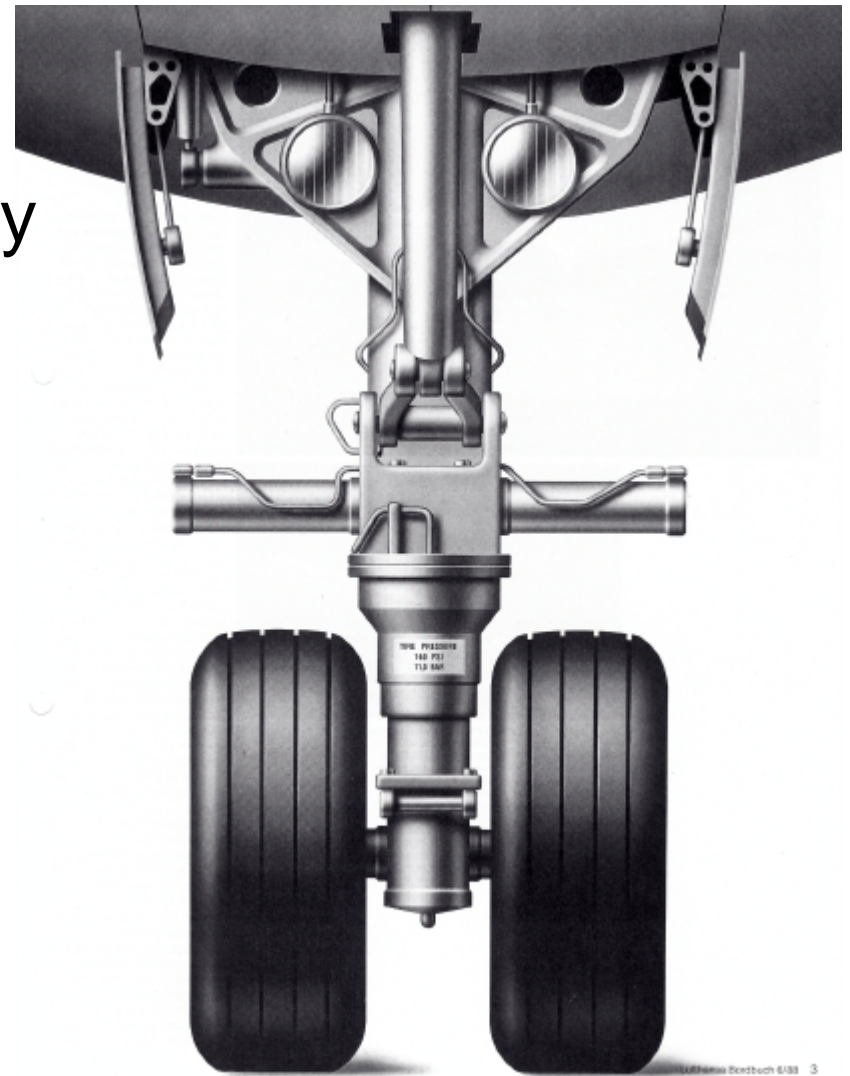
Airbus-Deutschland GmbH

Contents

- Review of Landing Gear History
- Struts and Shock Absorbers
- Tyres, Wheels & Brakes
- Landing Gear Systems
- Test & Certification

- Landing Gear Design
- Special Purpose Undercarriage

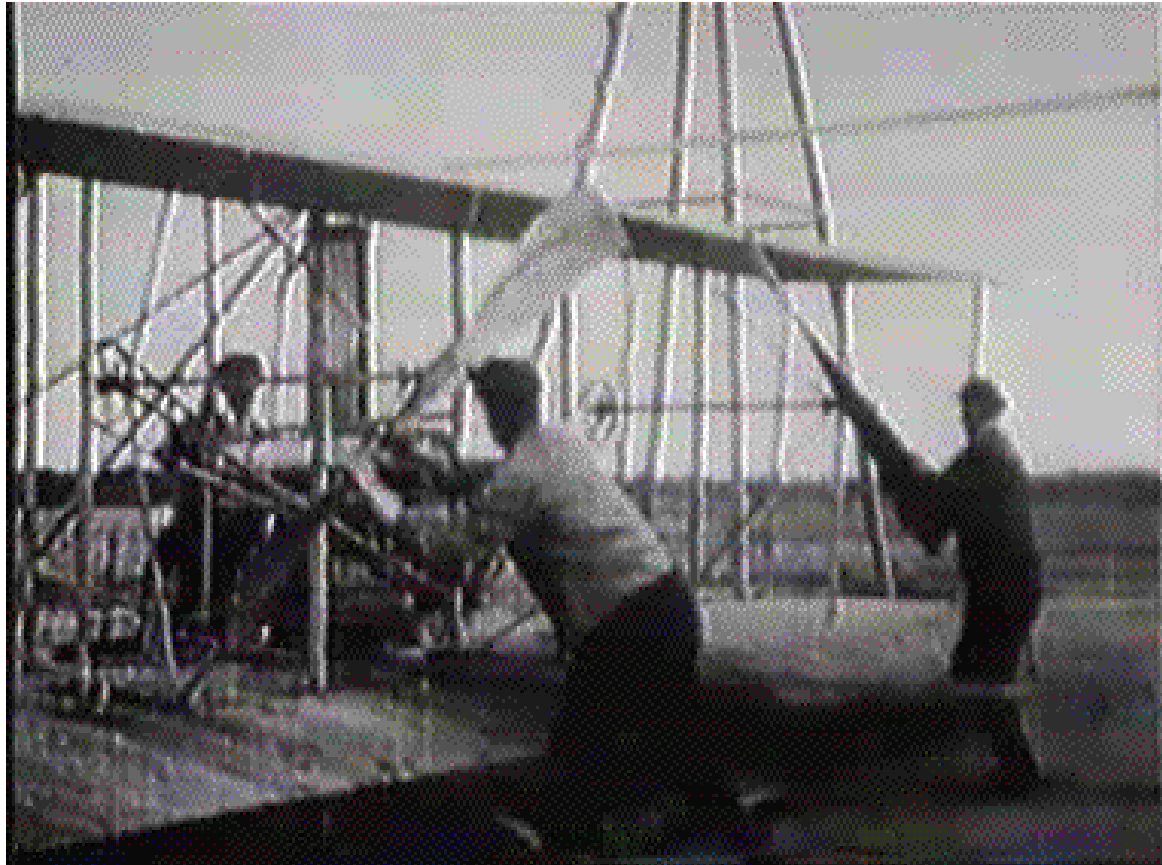
- Discussion



Landing Gear Data

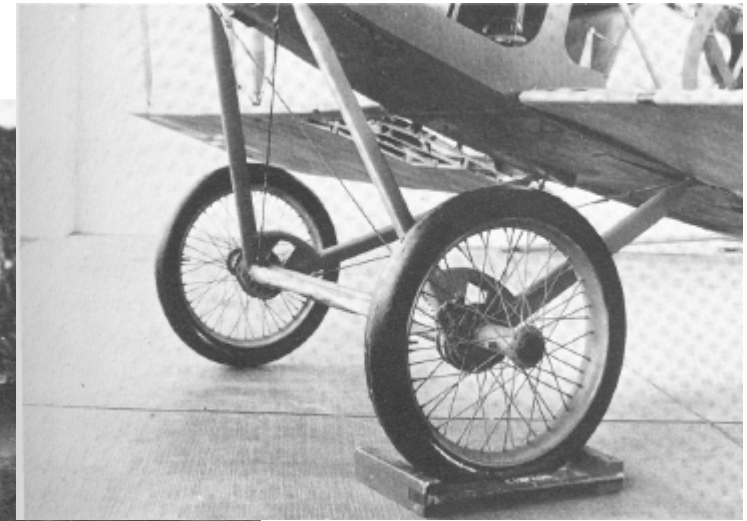
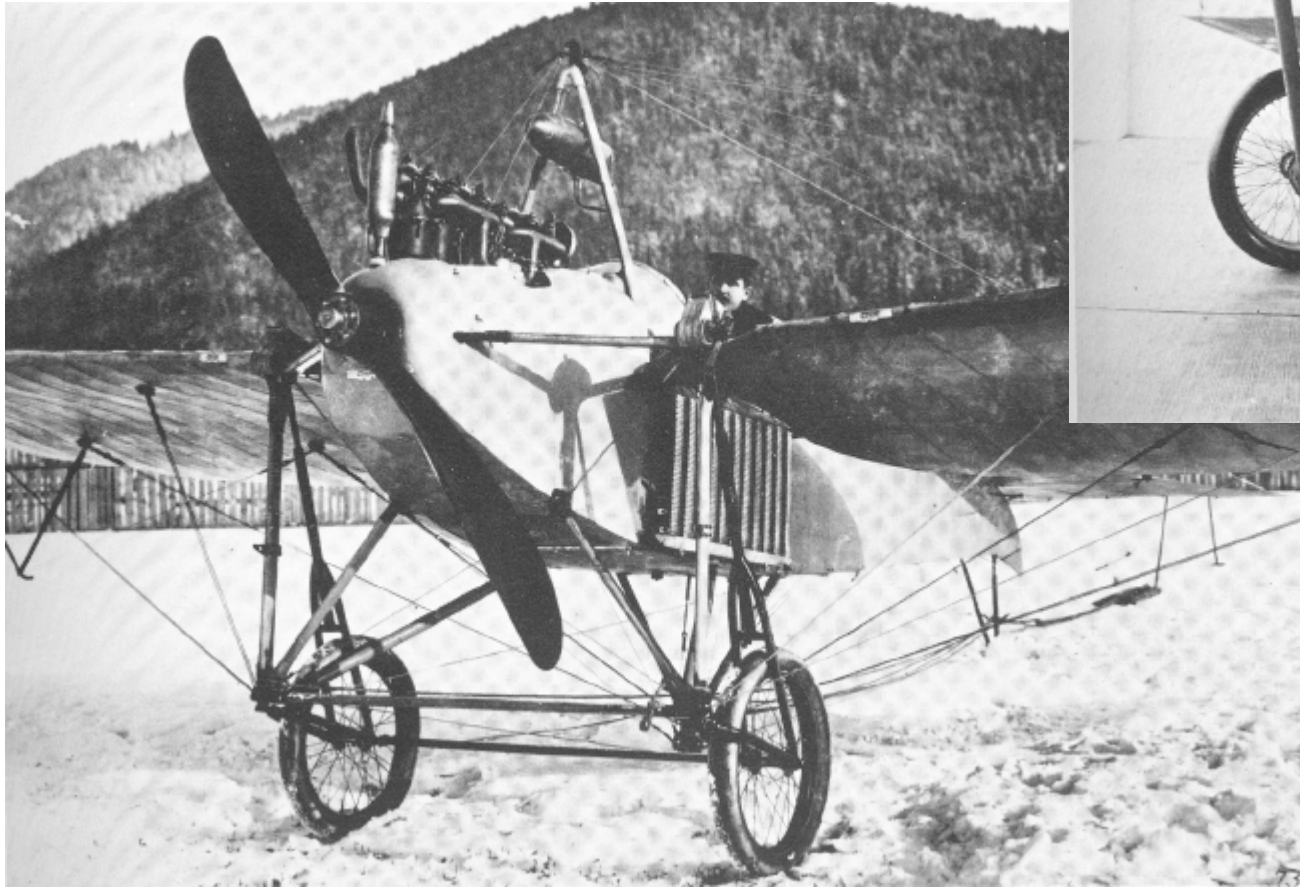
Weight:	2,5 ... 5 % of the MTOW
Cost:	1.5 to 1.75 % of the Aircraft
Maintenance:	20 % of the Airframe DMC
Loads:	up to 30,000 kg/wheel
Speeds:	over 300 km/h
Rolling distance:	up to 500.000 km
Life time of:	60.000 hours / 20 years
In-Service cycle:	20.000 hours (overhaul)

History



The Beginning - Wright Brothers , 1903

History



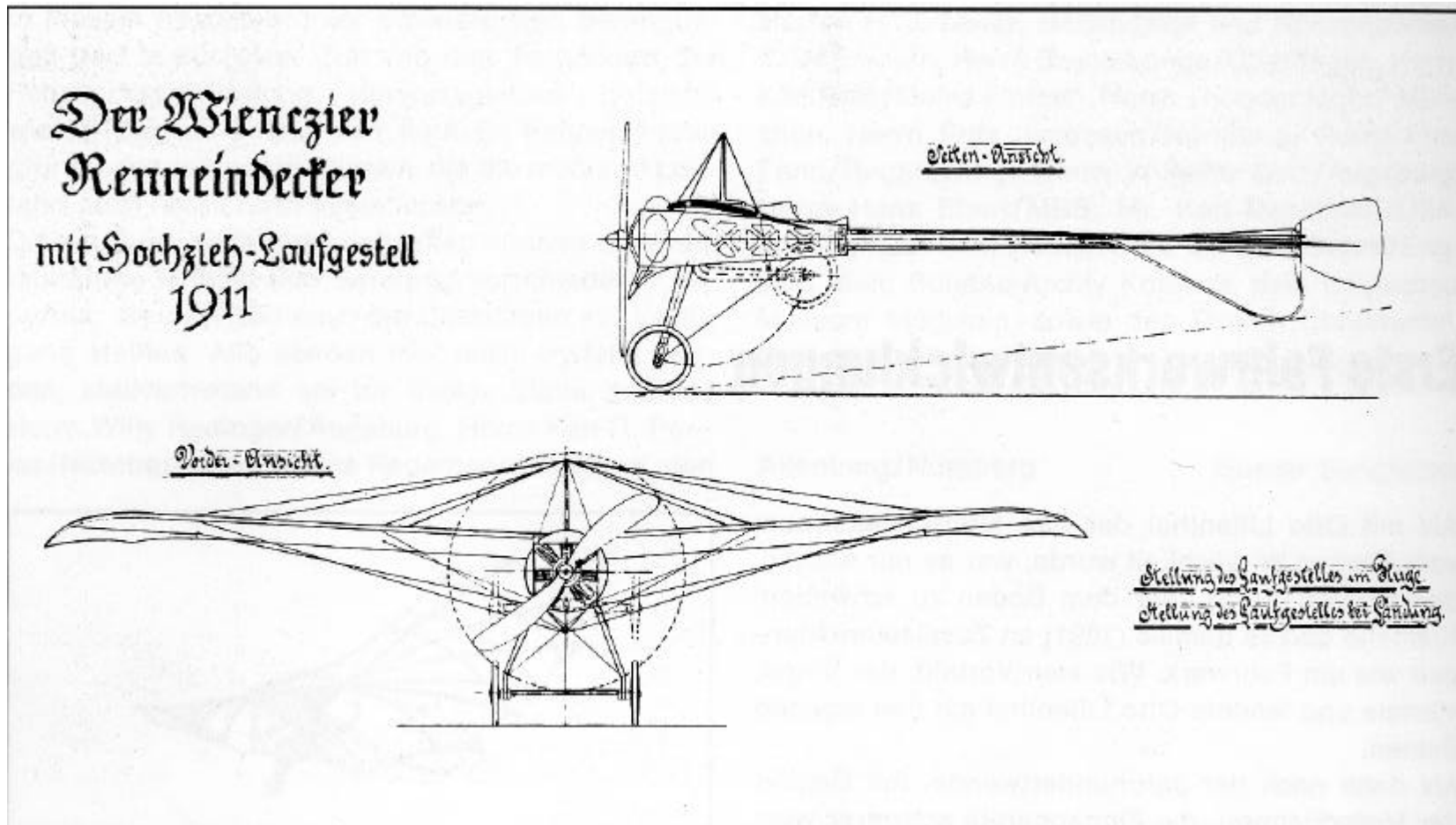
Etrich , 1914 / Albatros , 1918

History



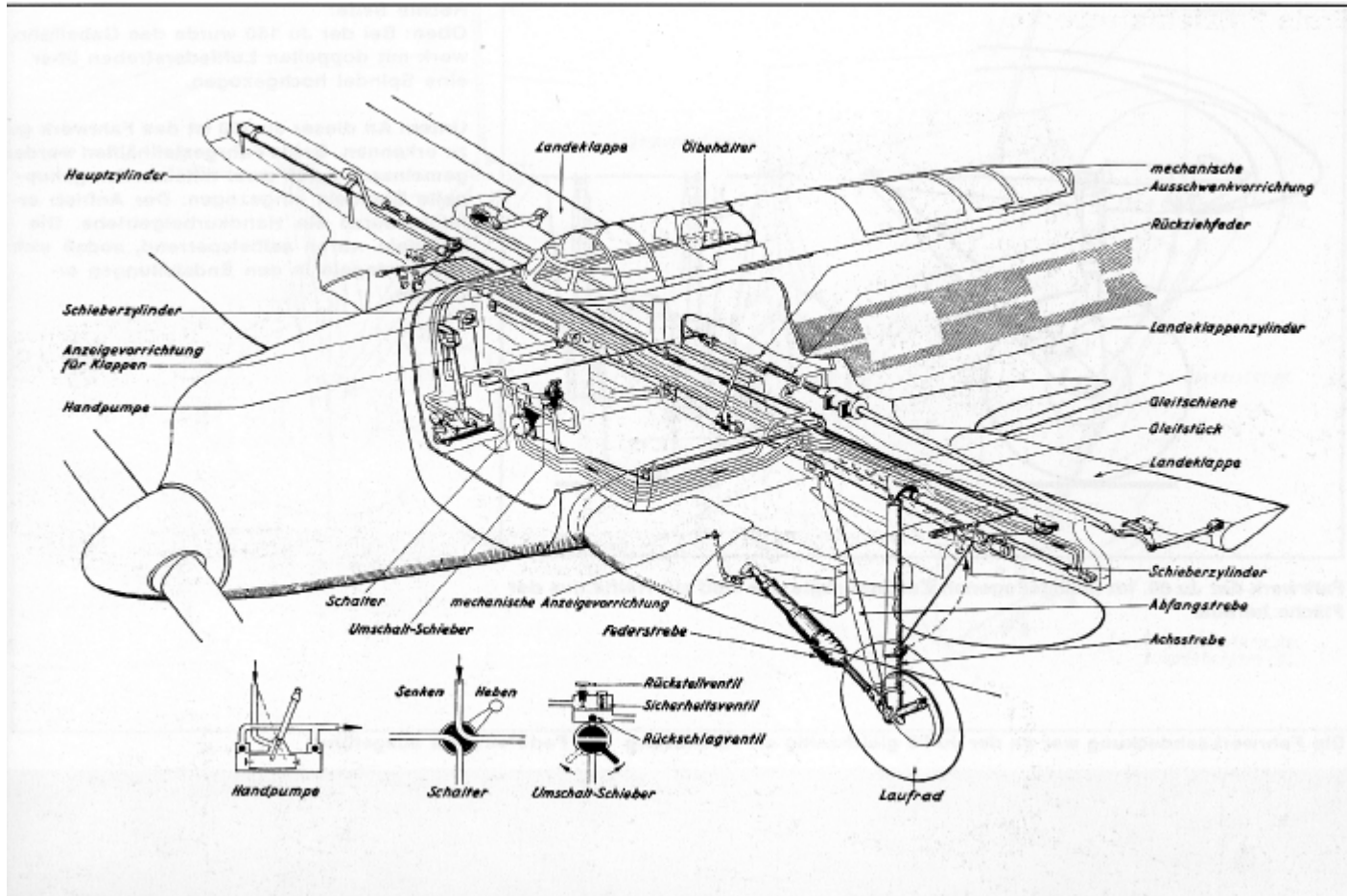
Messerschmitt Me 20 , 1928

History



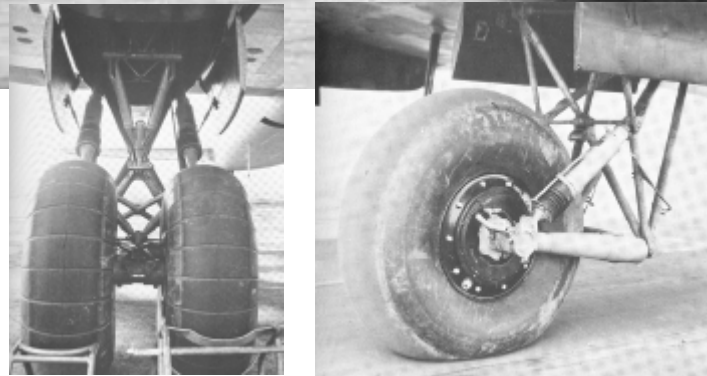
Retractable Landing Gear , 1911

History



Hydraulic powered Landing Gear , He-70 , 1932

History



Tail Wheel - Type , Focke Wulf Fw 200 , 1936

History



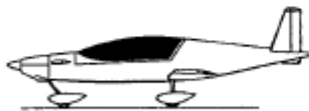
Nose wheel – type
(retractable)



Tail wheel – type
(retractable)



Tandem – type
(retractable)



Nose wheel – type
(fixed)



Tail wheel – type
(fixed)

Overview of Landing Gear Types

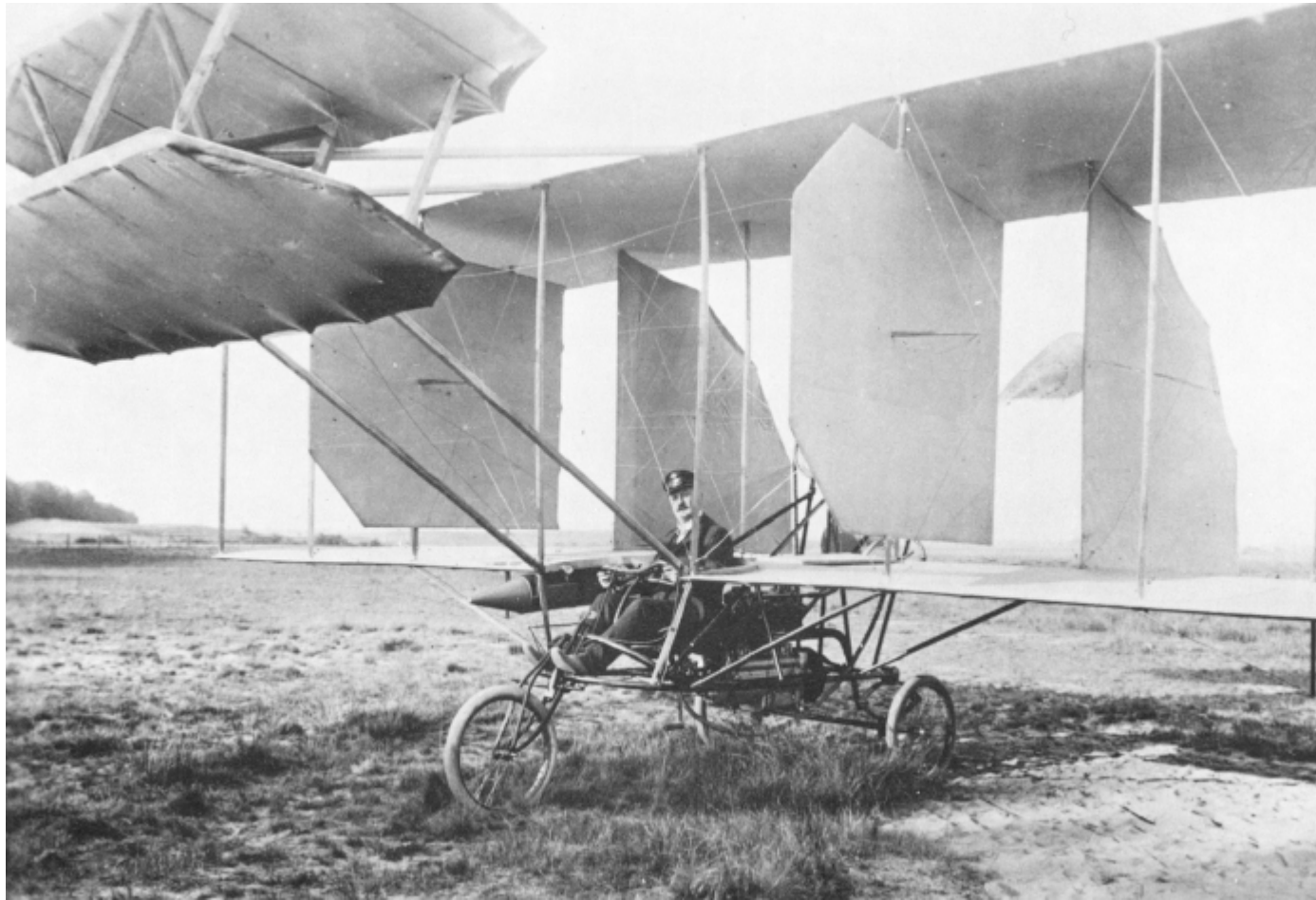
History



Advantages of the NLG-Type:

- **Improved rolling characteristics**
- **Horizontal floor**
- **The Pilot has better visibility**
- **Easier ground handling (e.g. towing)**

History



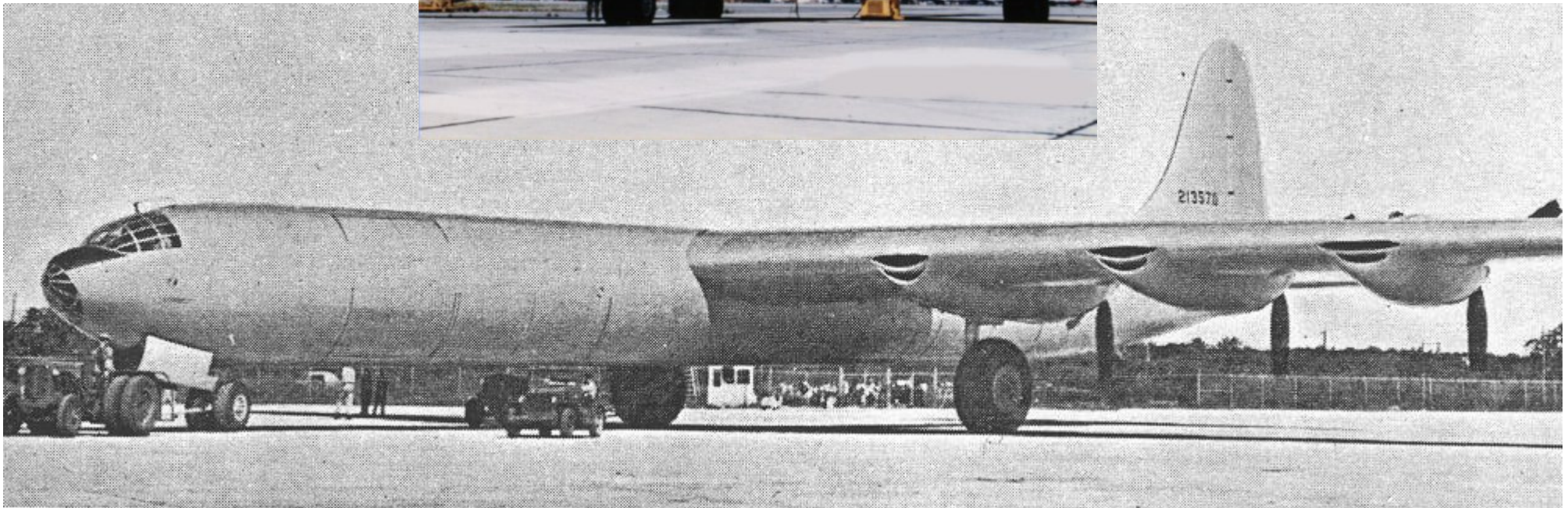
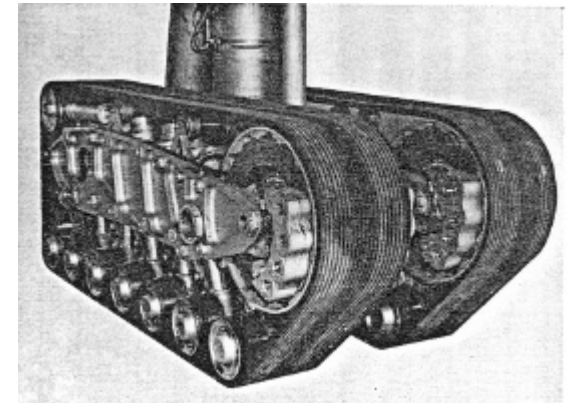
Nose Wheel-Type Landing Gear , Jatho IV , 1908

History



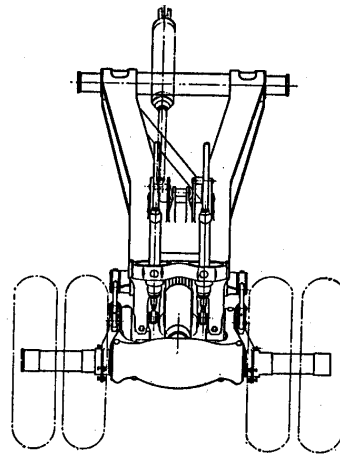
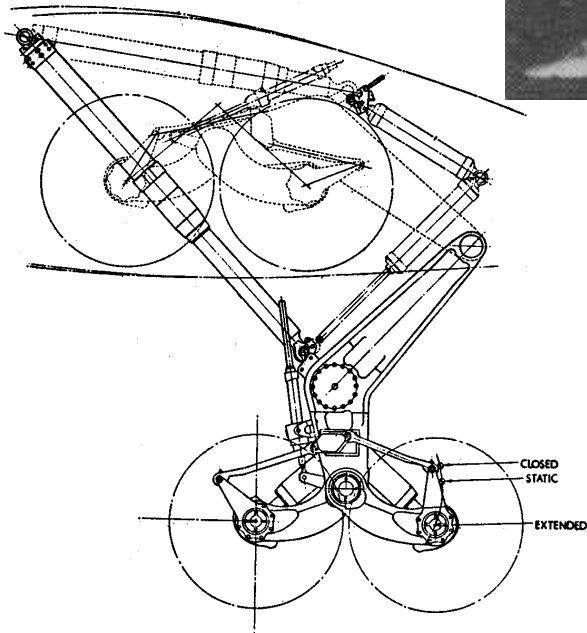
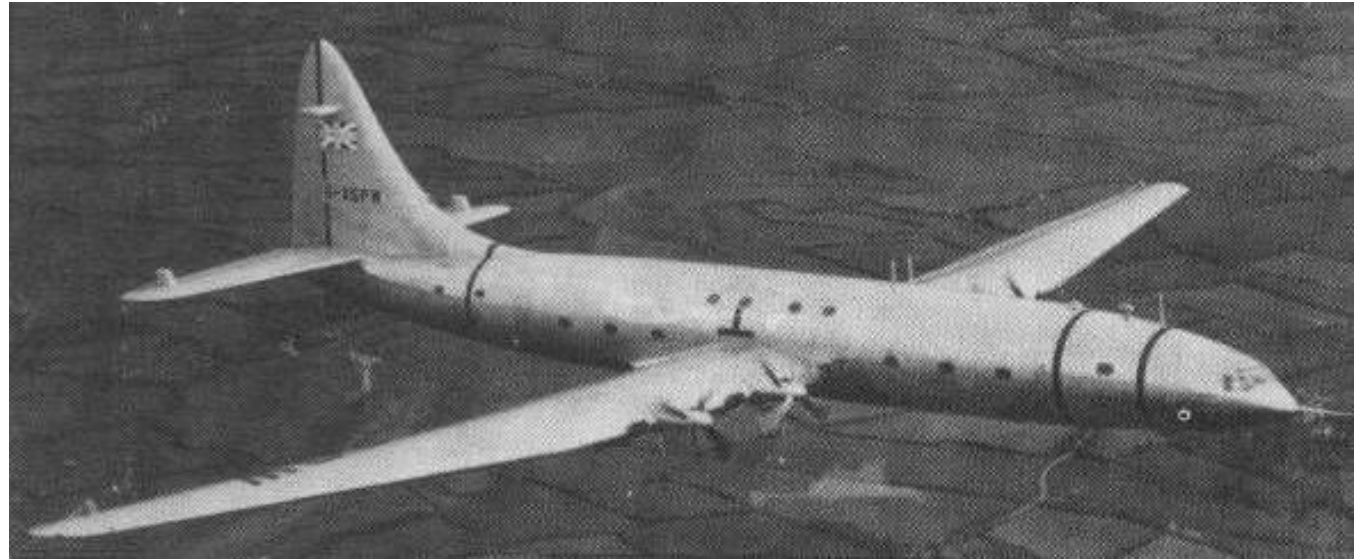
NLG-Type with Twin Wheels , B-29 , 1939

History



Single Wheels , Convair XB-36 , 1946

History



8-Wheel MLG , Bristol Brabazon , 1946

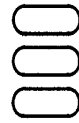
History



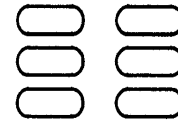
Single



Tandem



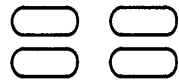
Triple



Triple Tandem



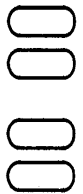
Twin



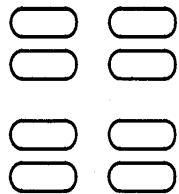
Twin Tandem



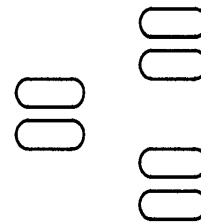
Tri-Twin Tandem



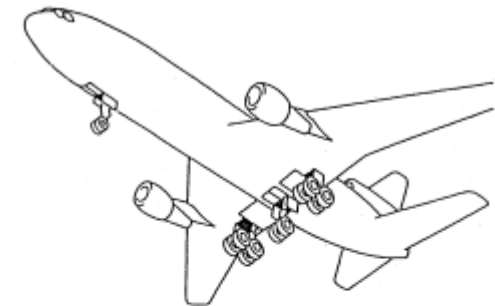
Dual Twin



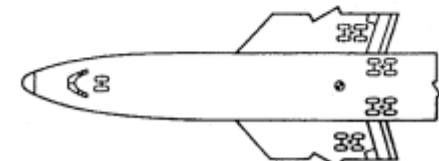
Dual Twin Tandem



Twin Tricycle



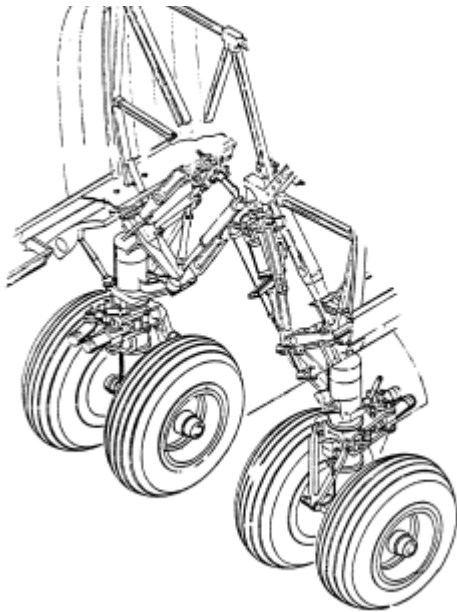
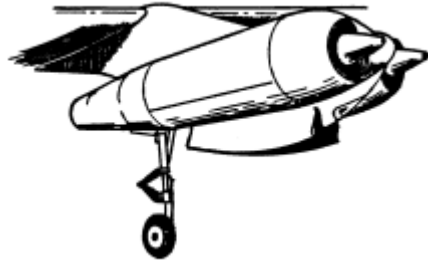
DC-10



B 747-400

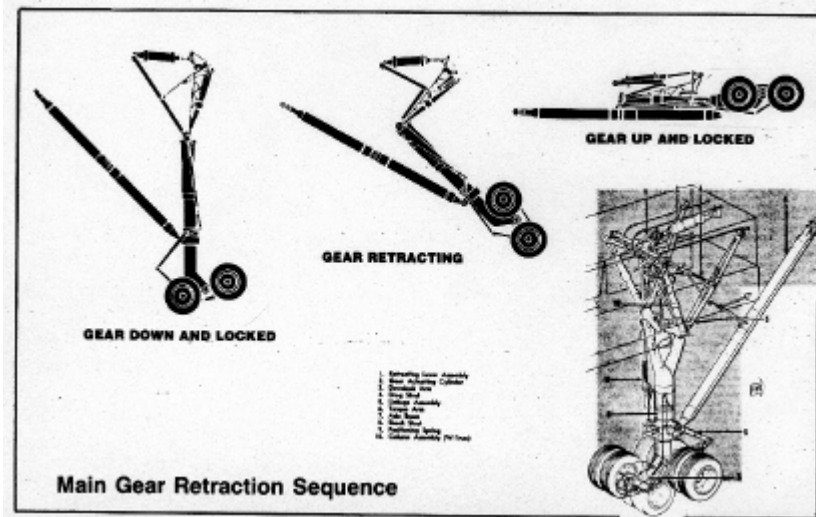
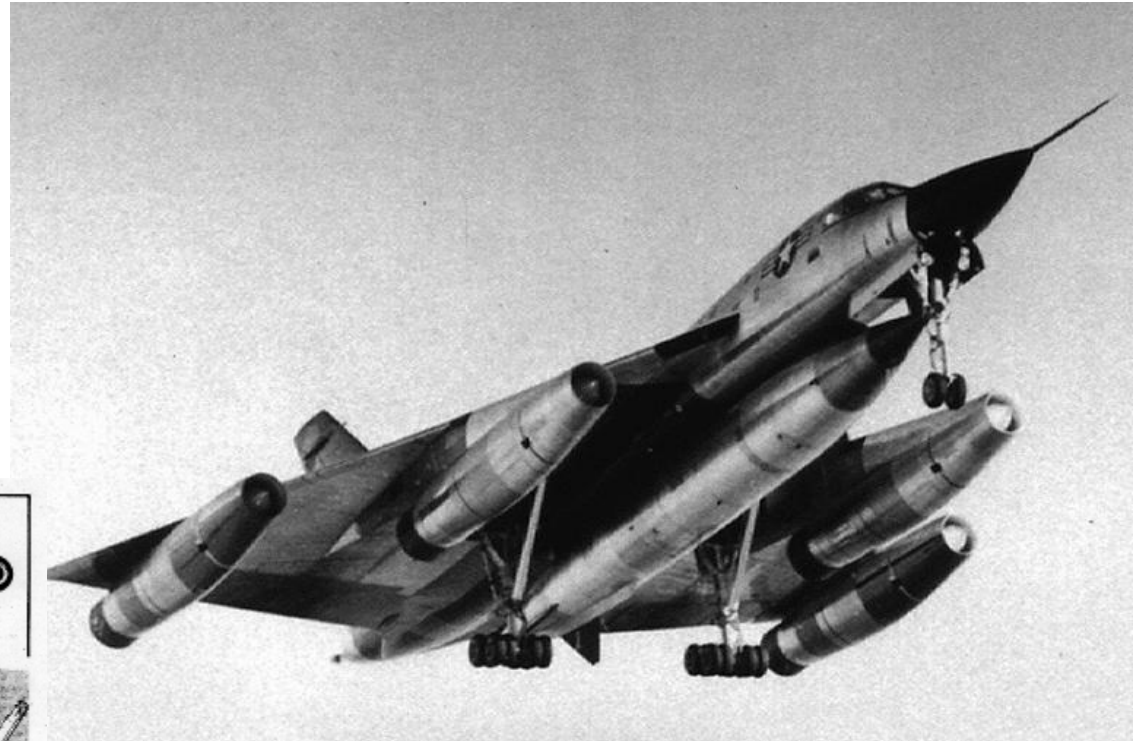
Overview Wheel Arrangements

History



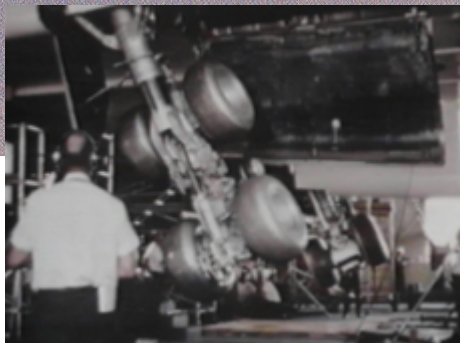
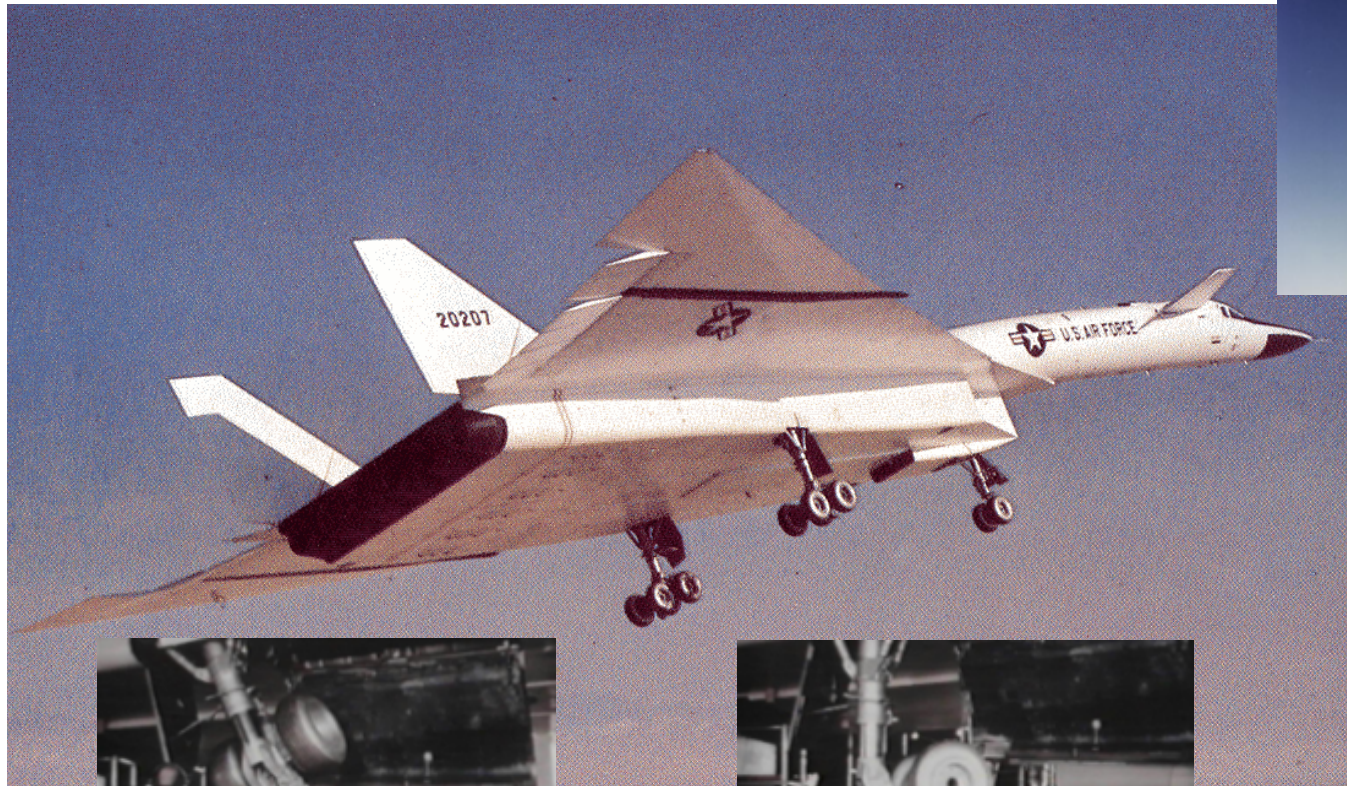
Tandem Gear , Boeing B-52 , 1954

History



Convair B-58 , 1956

History

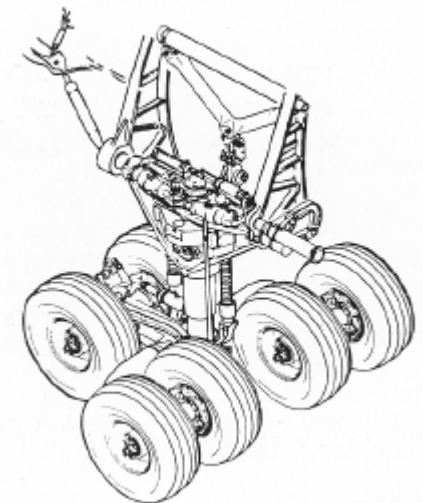
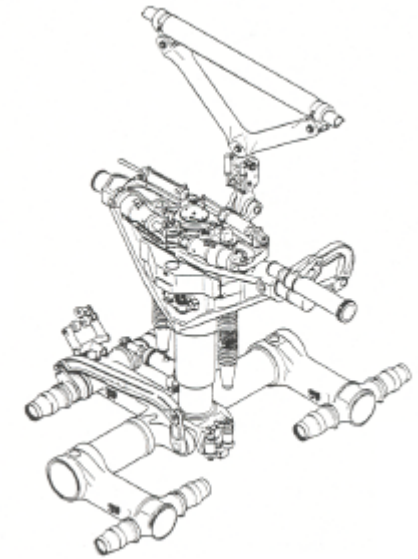


North American XB-70 , 1964

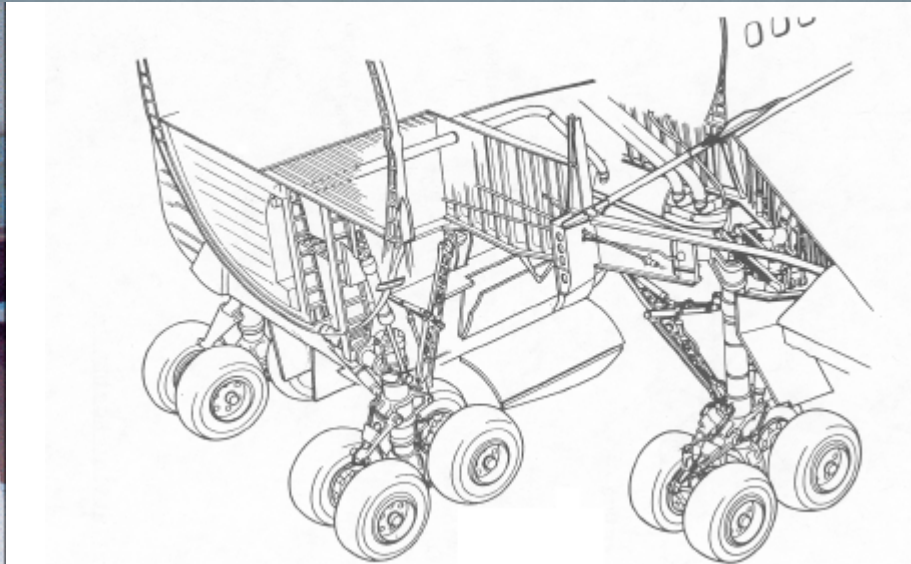
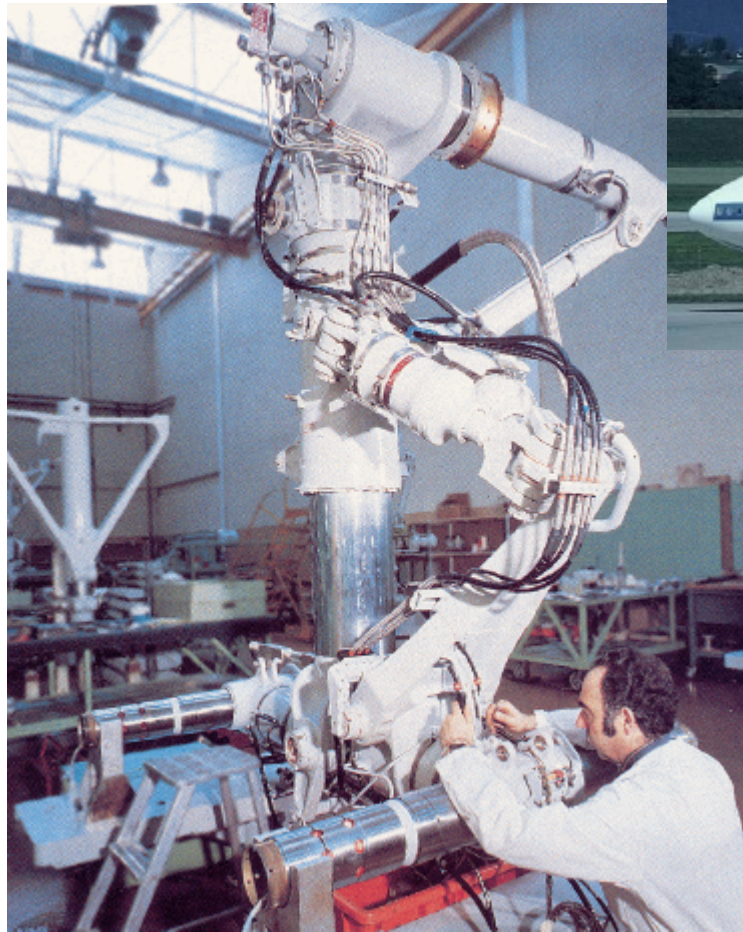
History



Lockheed C-5 Galaxy , 1968

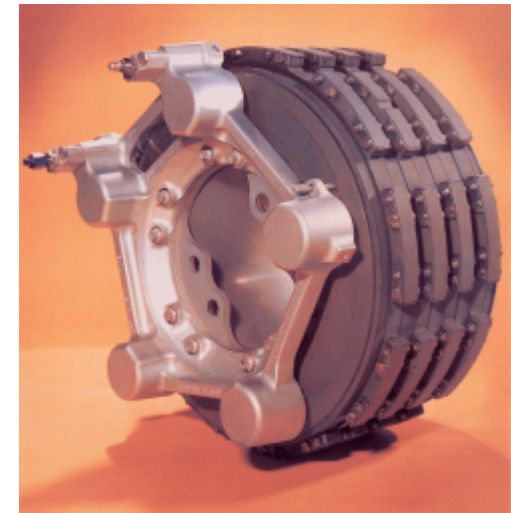
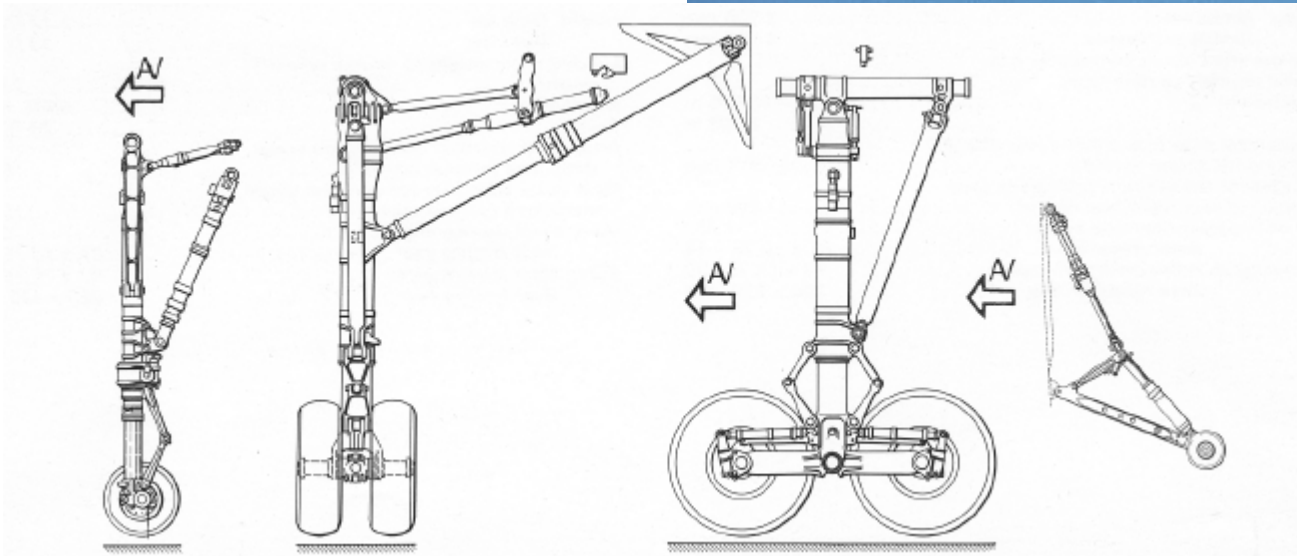


History



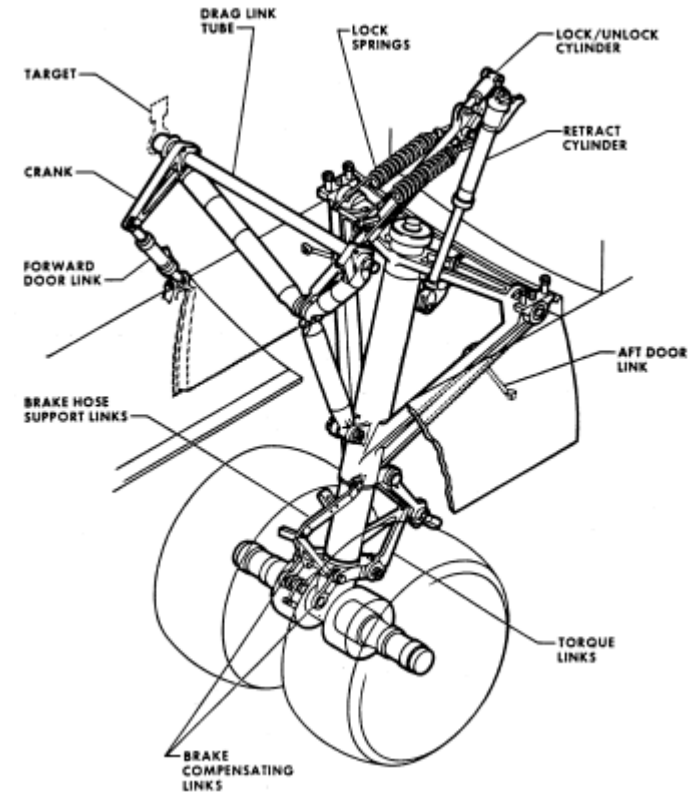
Boeing B747-100 , 1969

History



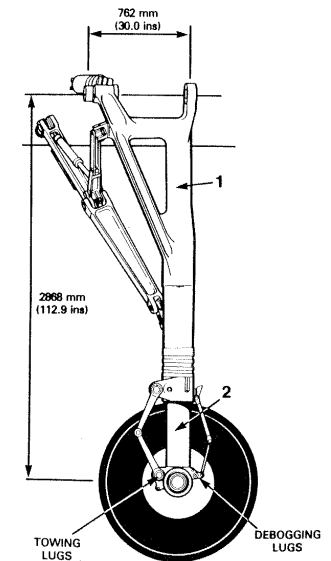
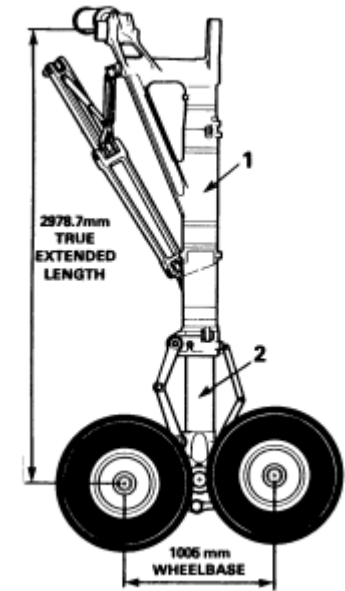
Concorde , 1969

History



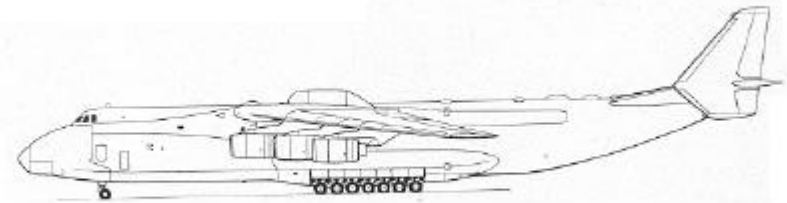
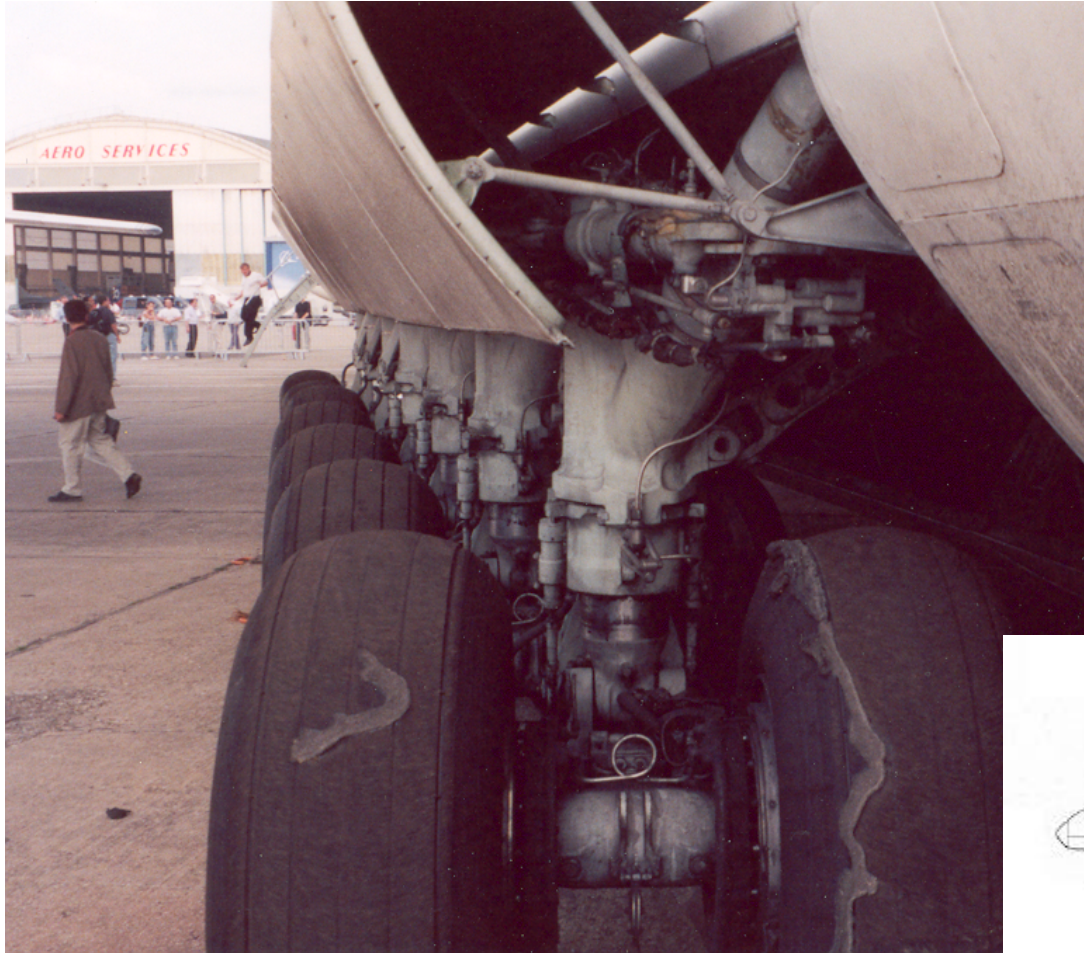
McDonnell Douglas DC-10-30 , 1972

History



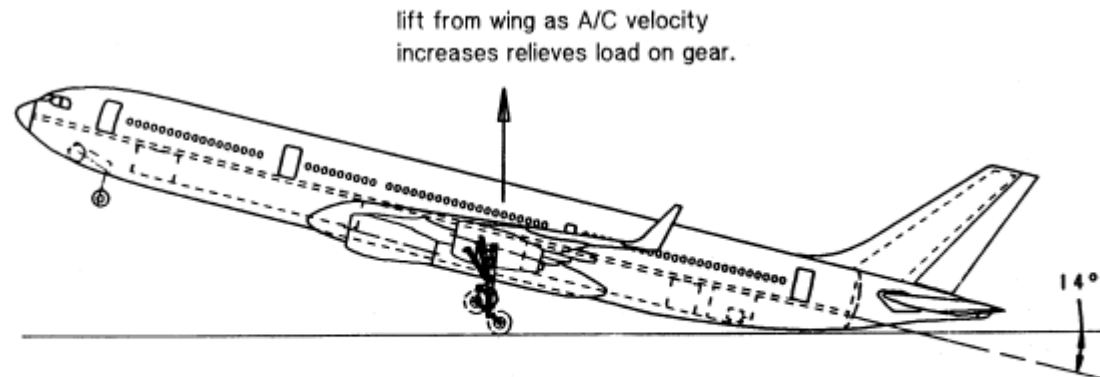
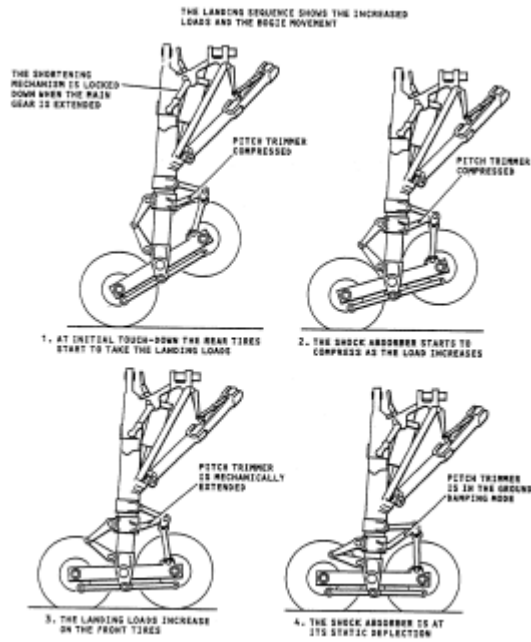
Airbus A320 , 1987

History



Antonow An-225 , 1988

History



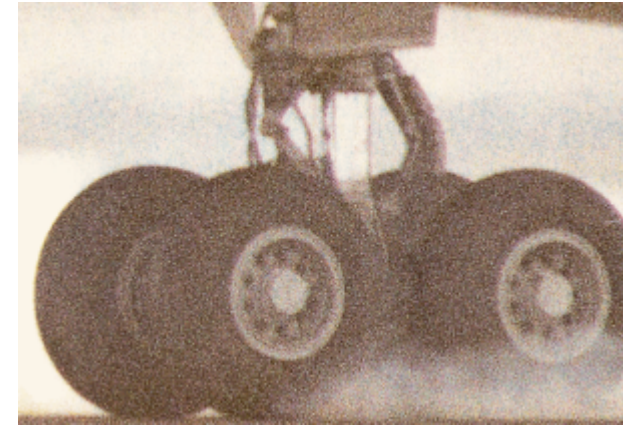
During take-off main gear acts as levered undercarriage.
 As load is relieved from main gear the shock absorber extends and the bogie beam rotates about forward wheels due to the geometry of the articulation links and pitch trimmer.
 A/C rotates about rear wheels only at take-off.

Airbus A340 , 1991

Landing Gear

General Requirements:

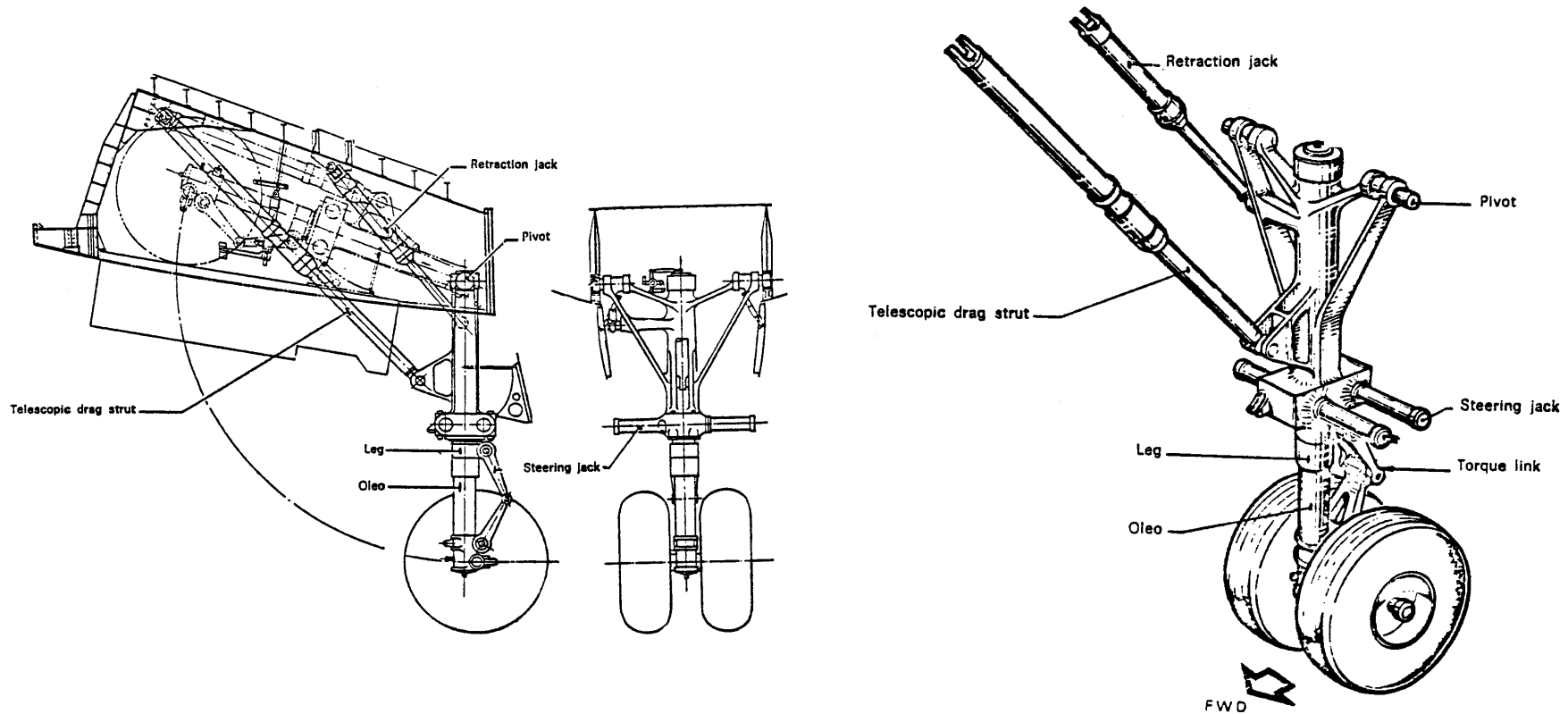
- Comfortable shock absorption
- Short stopping distance (braking)
- High rolling stability during ground manoeuvring
- Small storage volume
- Low drag (specially for fixed landing gears)
- High reliability and safety
- Low maintenance
- Low weight



Struts and Shock Absorbers



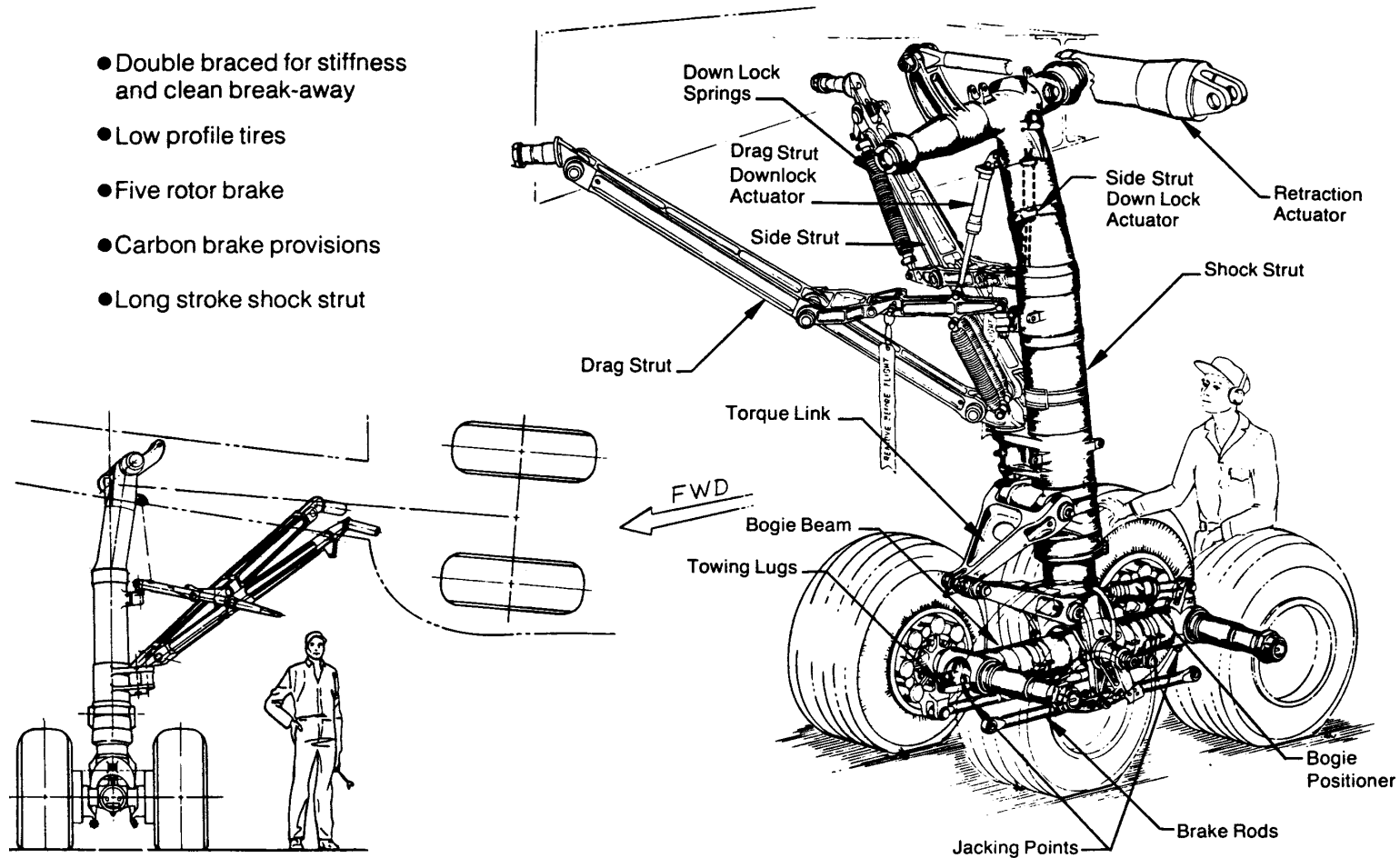
— Struts and Shock Absorbers



NLG (Nose Landing Gear) , Airbus A300

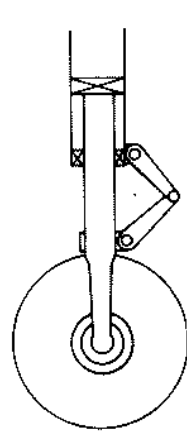
— Struts and Shock Absorbers

- Double braced for stiffness and clean break-away
- Low profile tires
- Five rotor brake
- Carbon brake provisions
- Long stroke shock strut

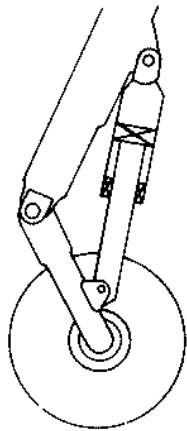


MLG (Main Landing Gear) , Boeing B767

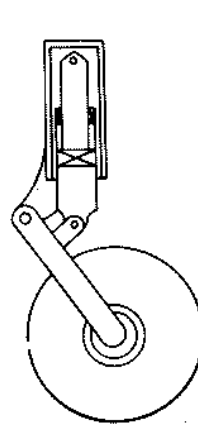
— *Struts and Shock Absorbers*



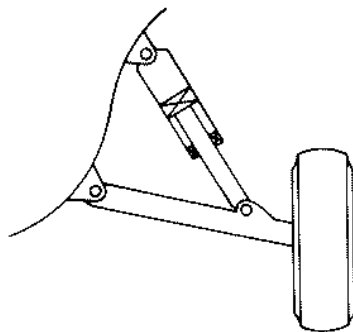
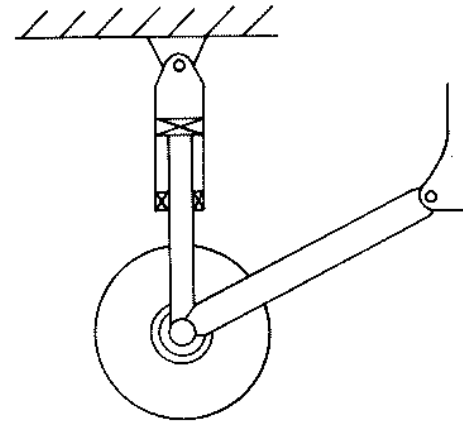
Telescopic



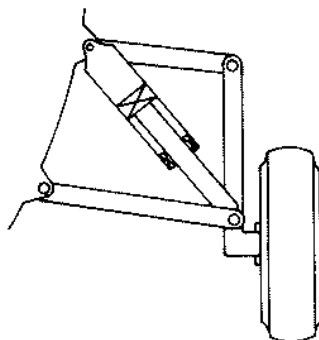
Lever suspension



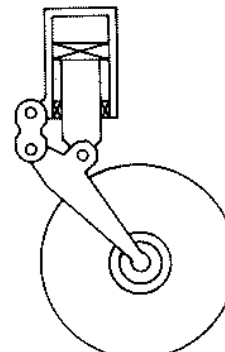
Forward lever



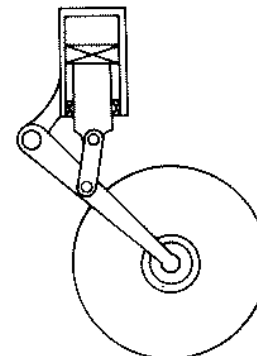
Sideways lever



Deformable quadrilateral

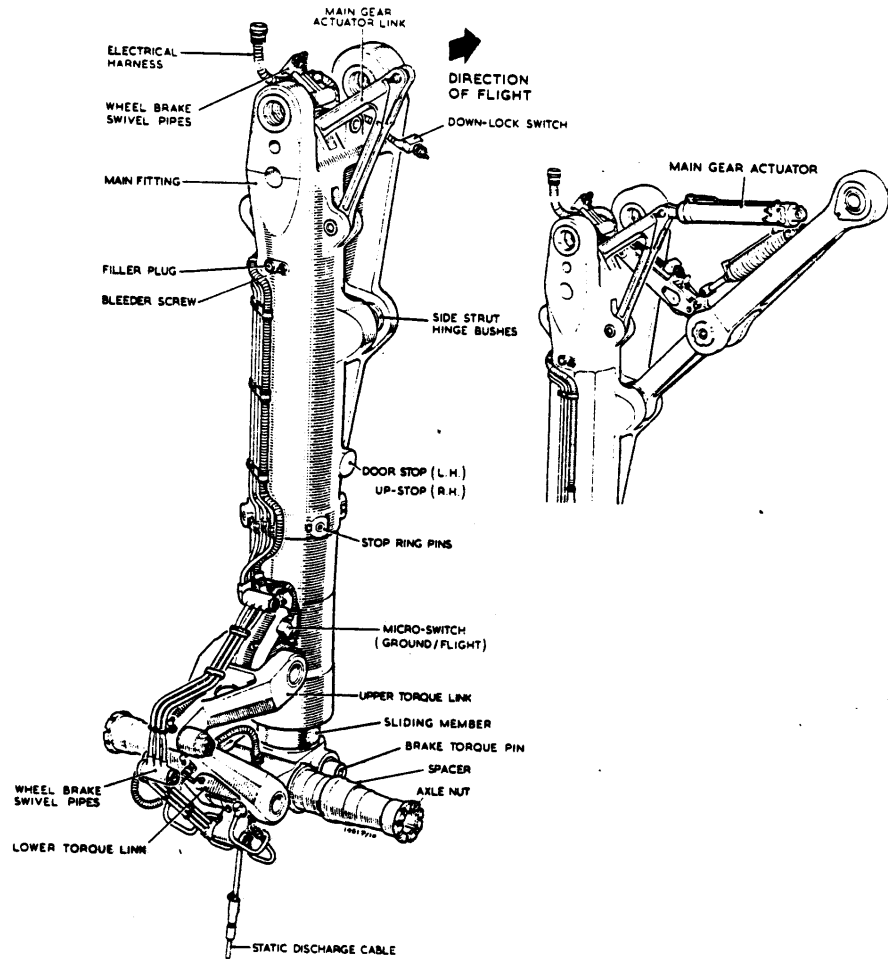


Semi-articulated



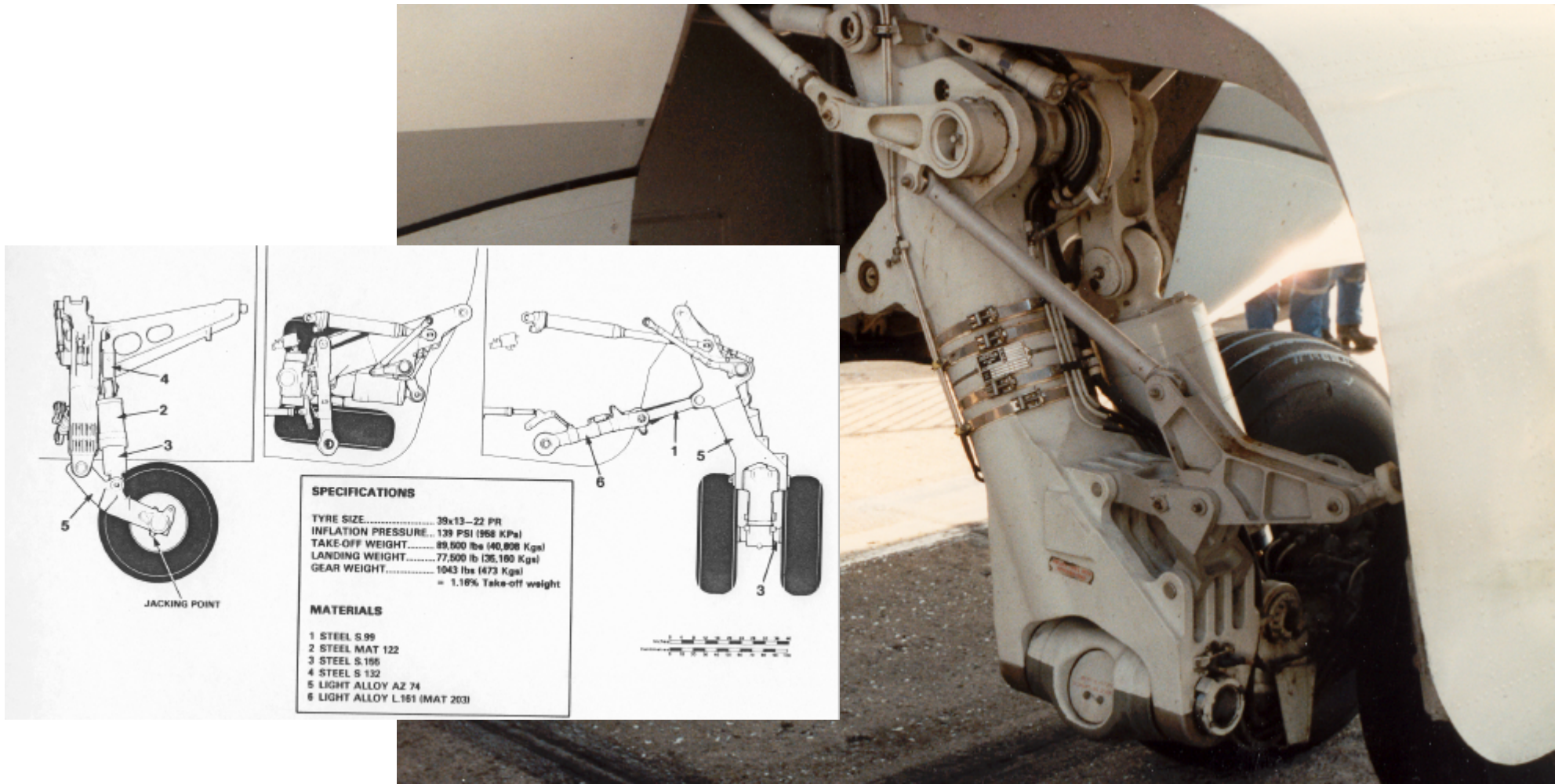
Overview Strut-Types

— Struts and Shock Absorbers



Telescopic-type MLG , Fokker F-28

— Struts and Shock Absorbers



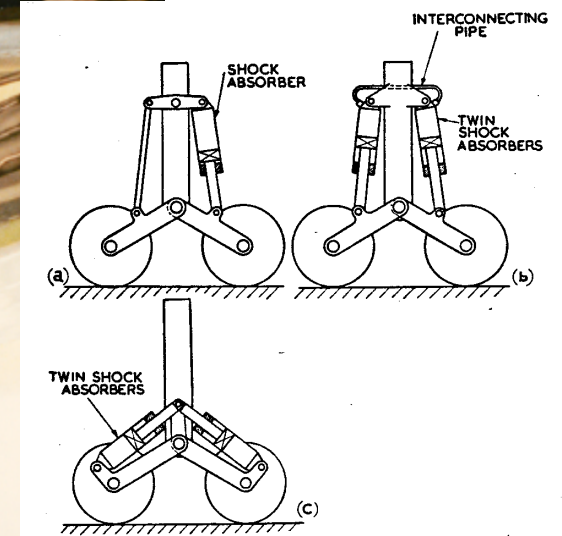
Lever-type MLG , BAe 146

Struts and Shock Absorbers



Semi-lever-type MLG , TU 154

— Struts and Shock Absorbers



4 Wheel MLG , Caravelle

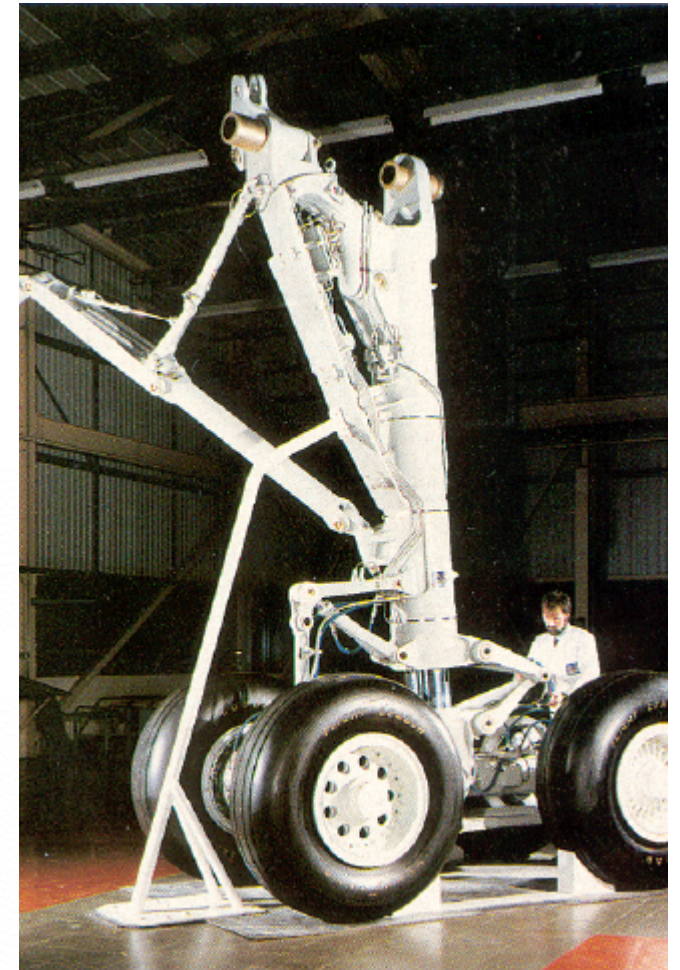
— *Struts and Shock Absorbers*



MLG , A300

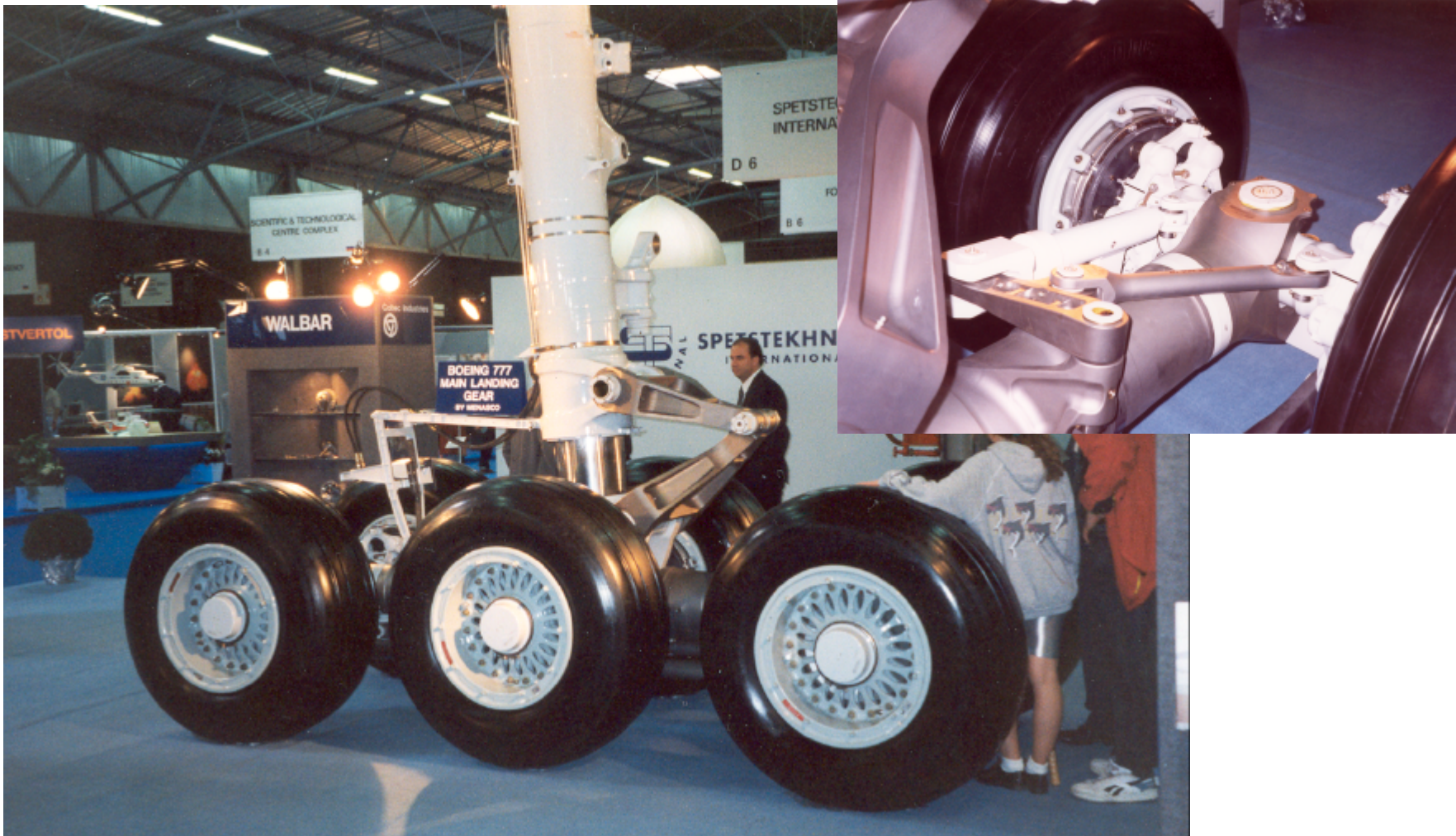


MLG , A310



MLG , A340

— *Struts and Shock Absorbers*



MLG with Steering , B777

Struts and Shock Absorbers



Aermacchi / Embraer AMX

Struts and Shock Absorbers



EF 2000 - Typhoon

Struts and Shock Absorbers



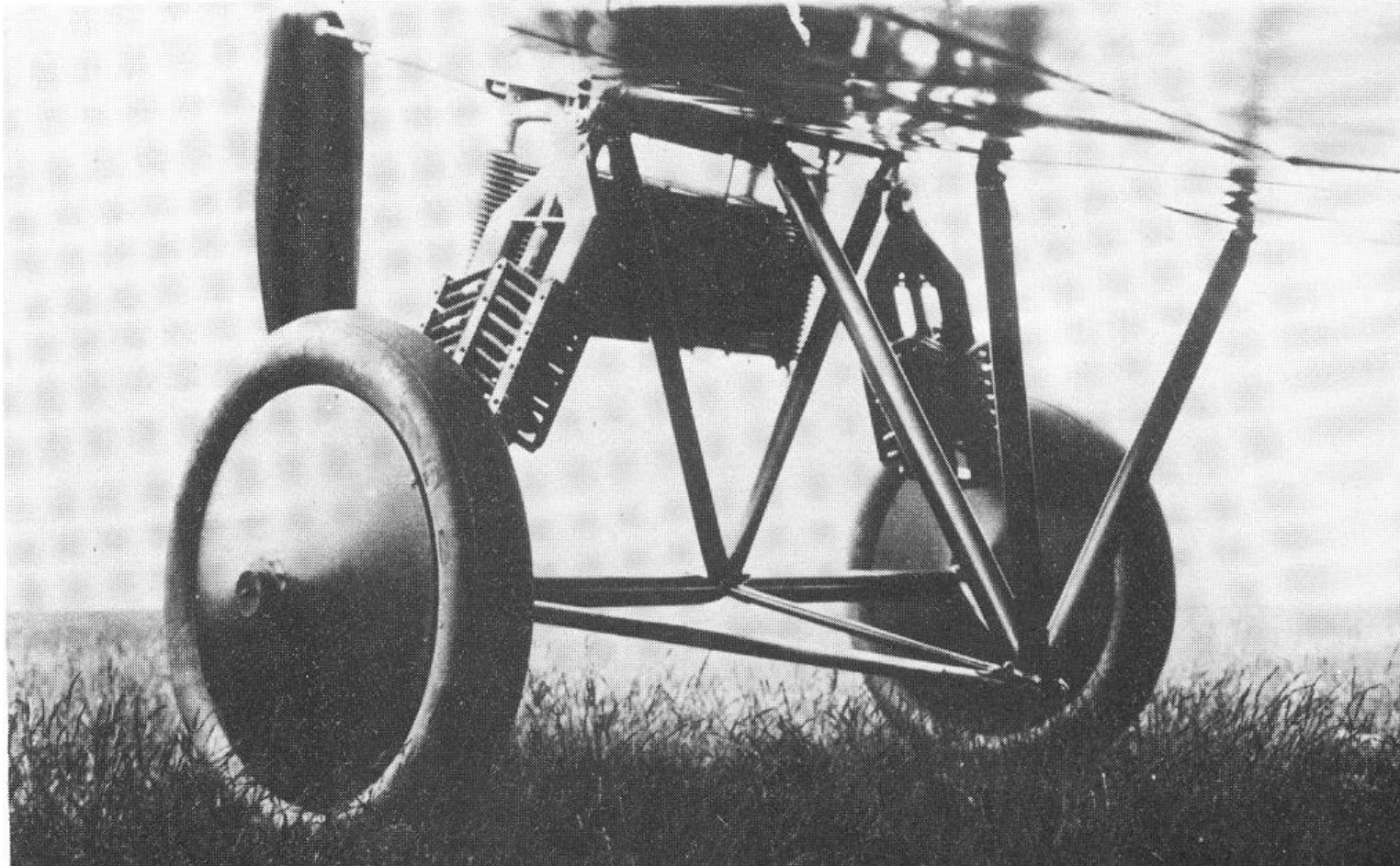
Lockheed Martin F-16

Struts and Shock Absorbers



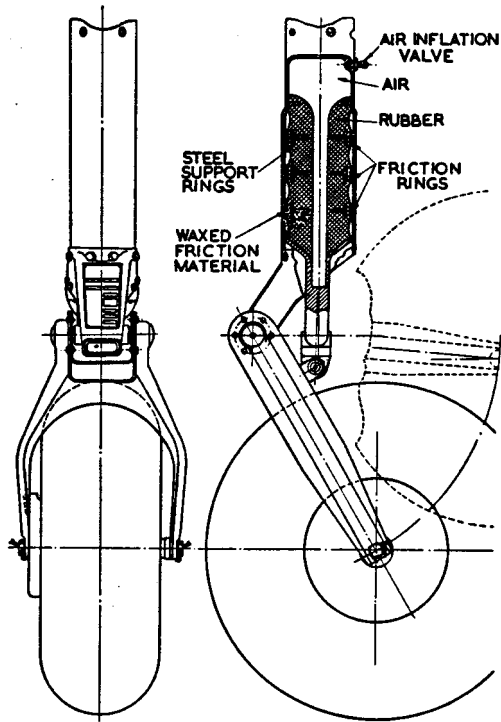
Boeing F-18

Struts and Shock Absorbers

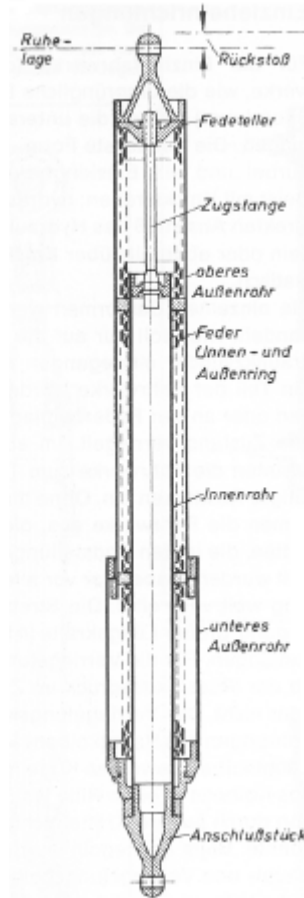


Rubber Shock Absorber

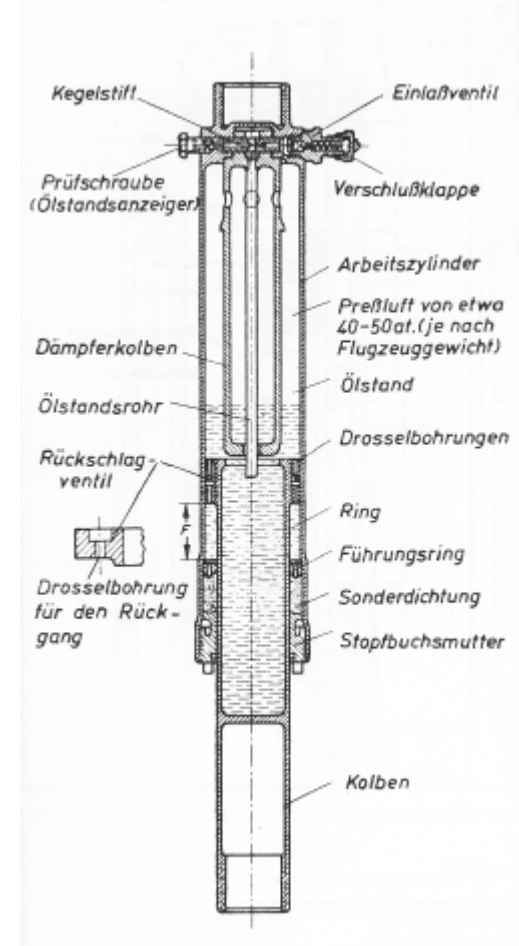
— Struts and Shock Absorbers



Rubber

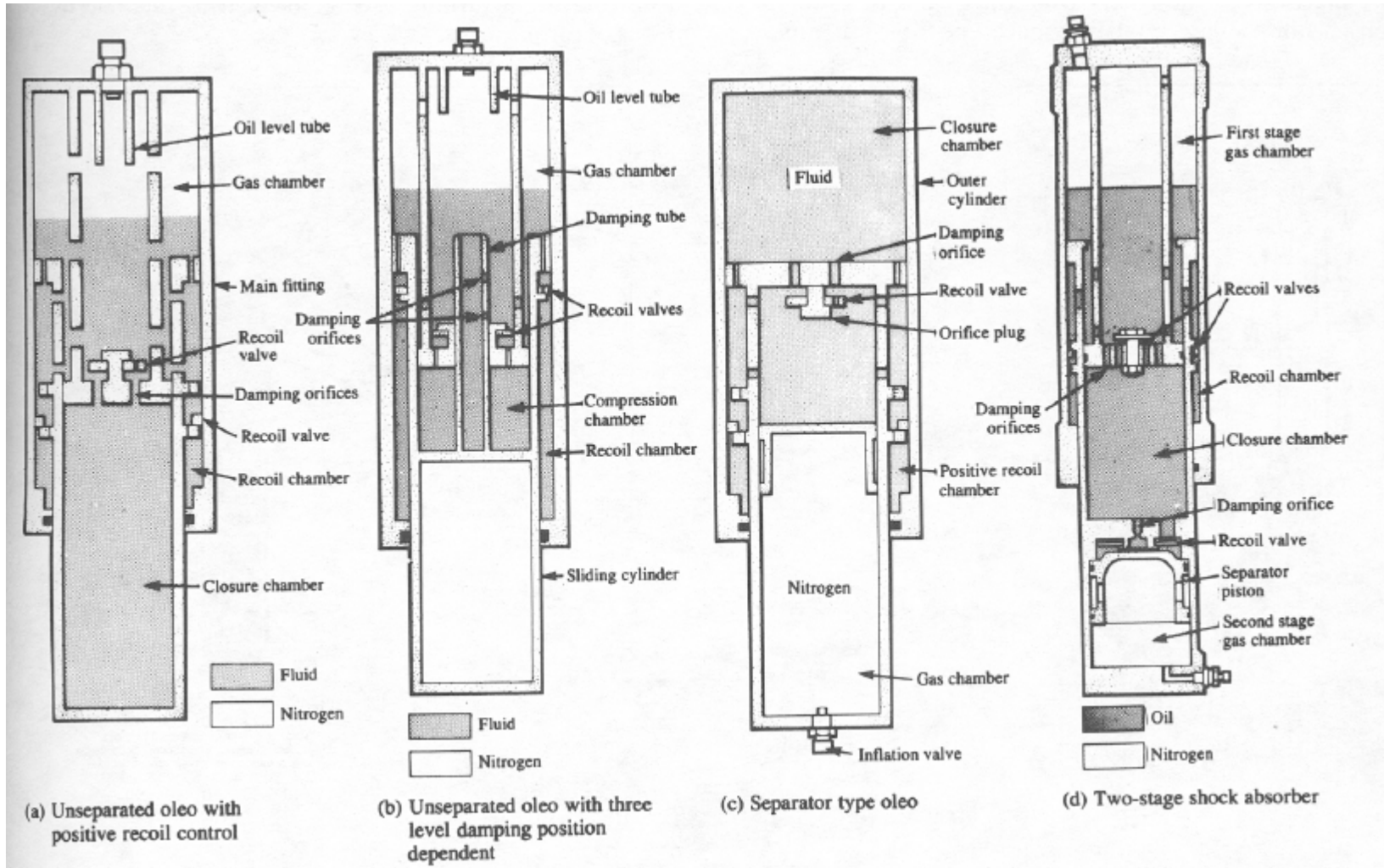


Spring



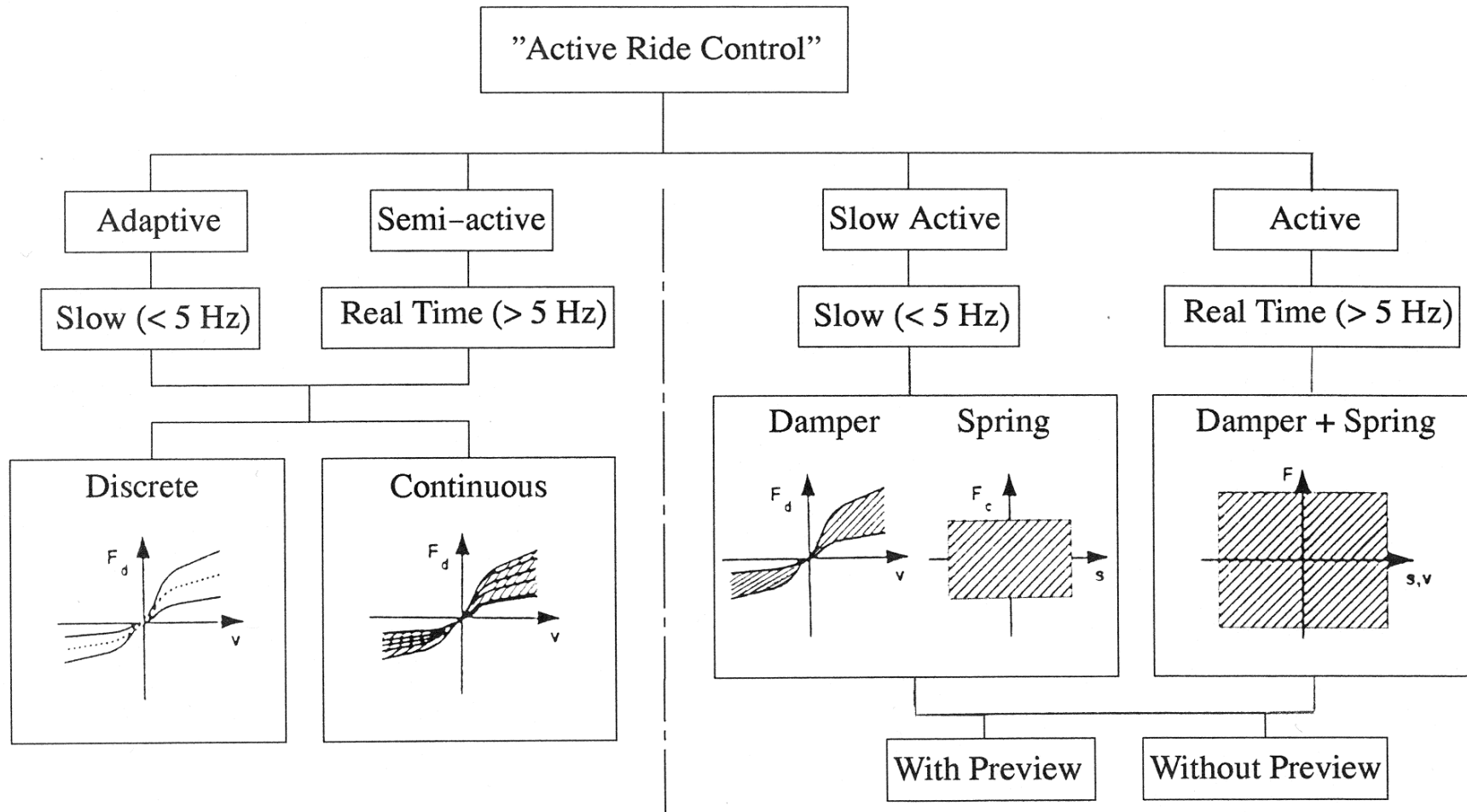
Oleo

— Struts and Shock Absorbers

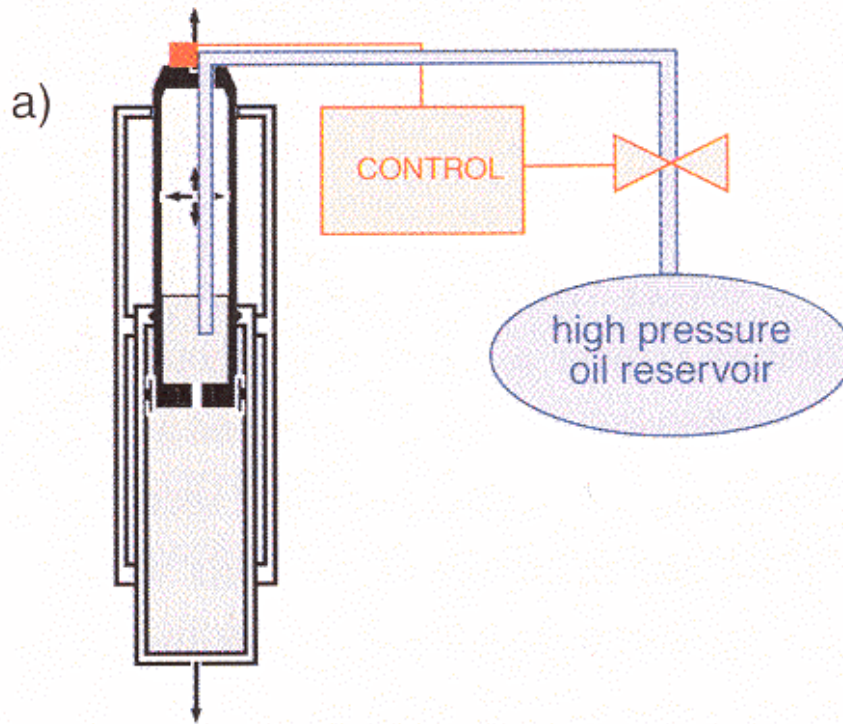


Oleo-Types

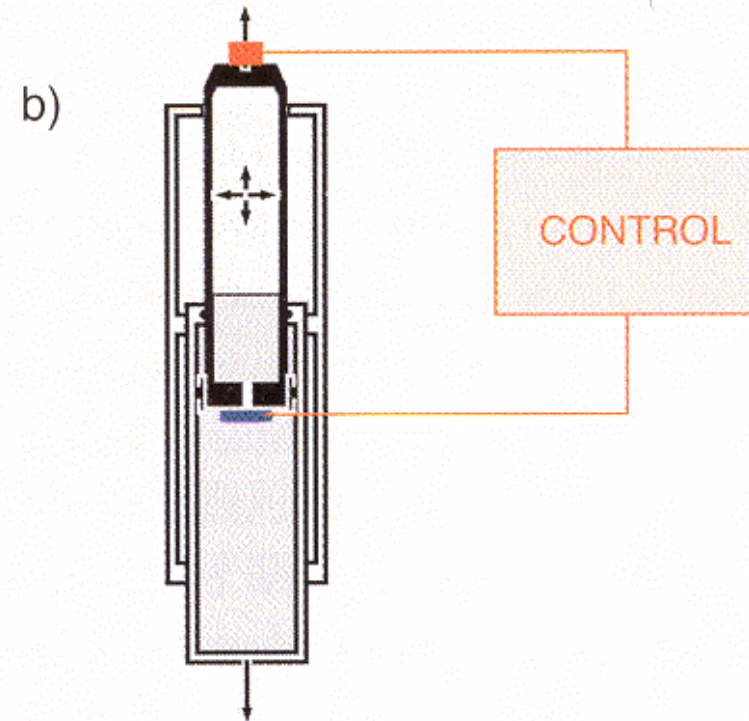
— Struts and Shock Absorbers



— *Struts and Shock Absorbers*

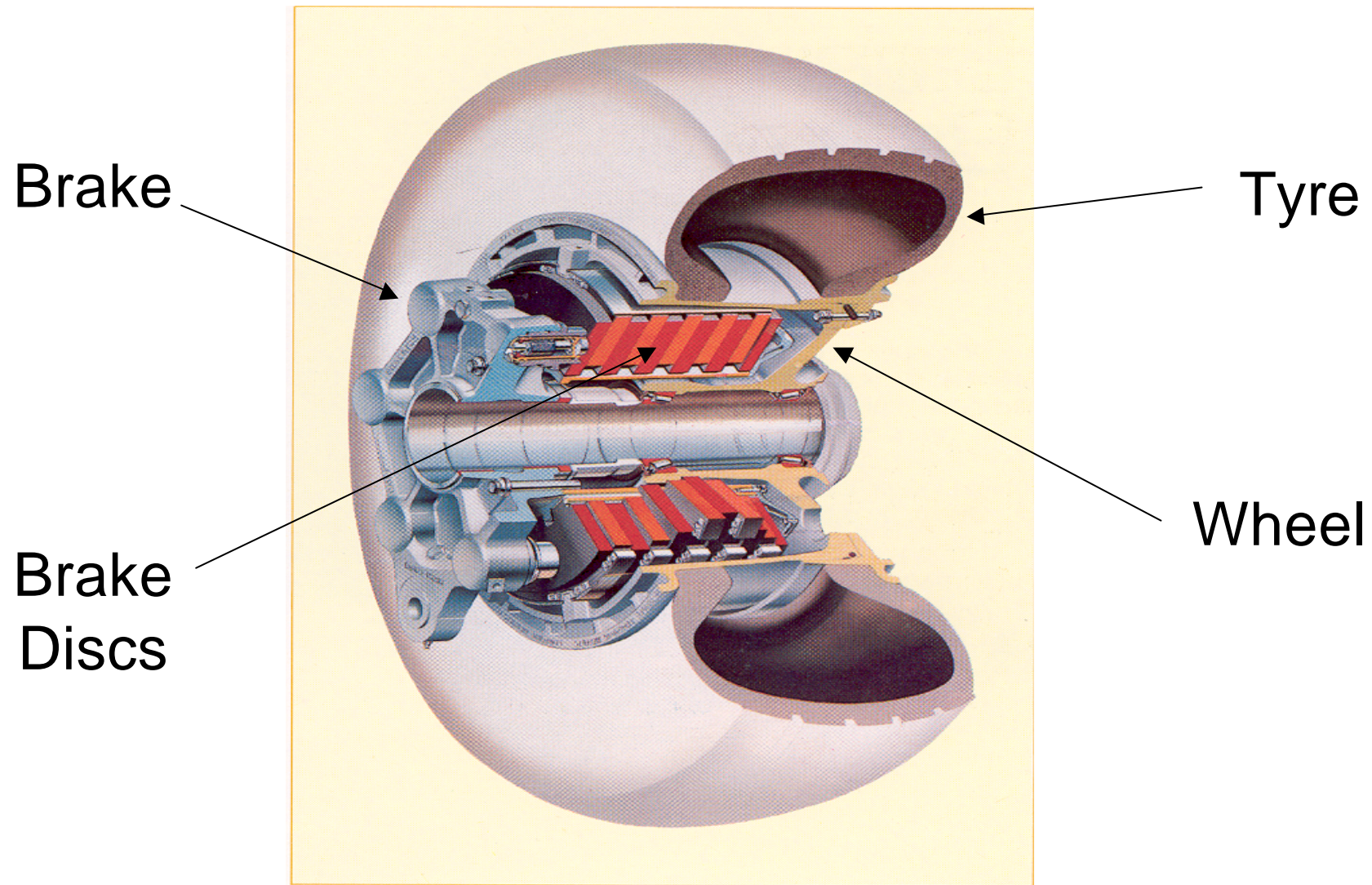


Active Control

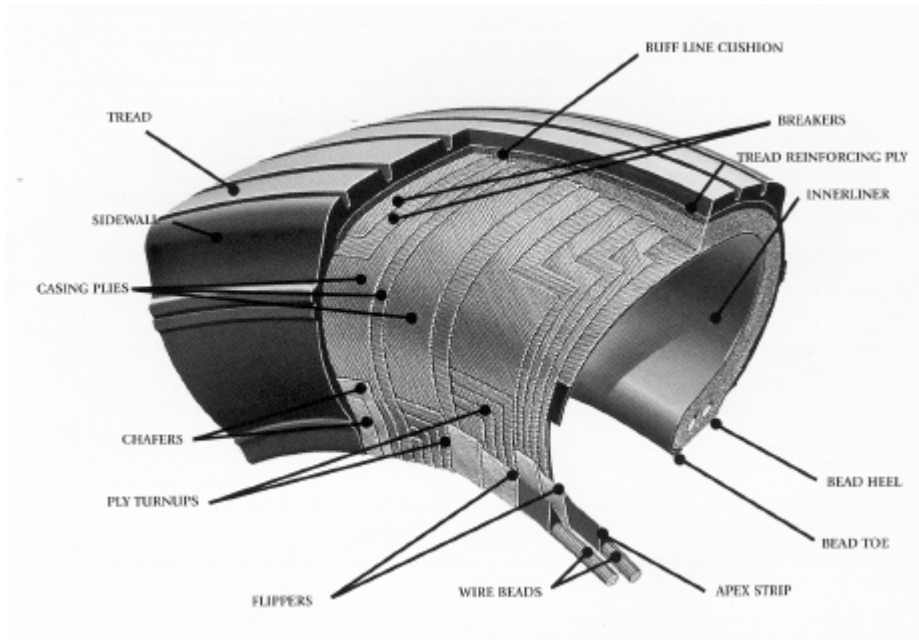


Semi-active Damping

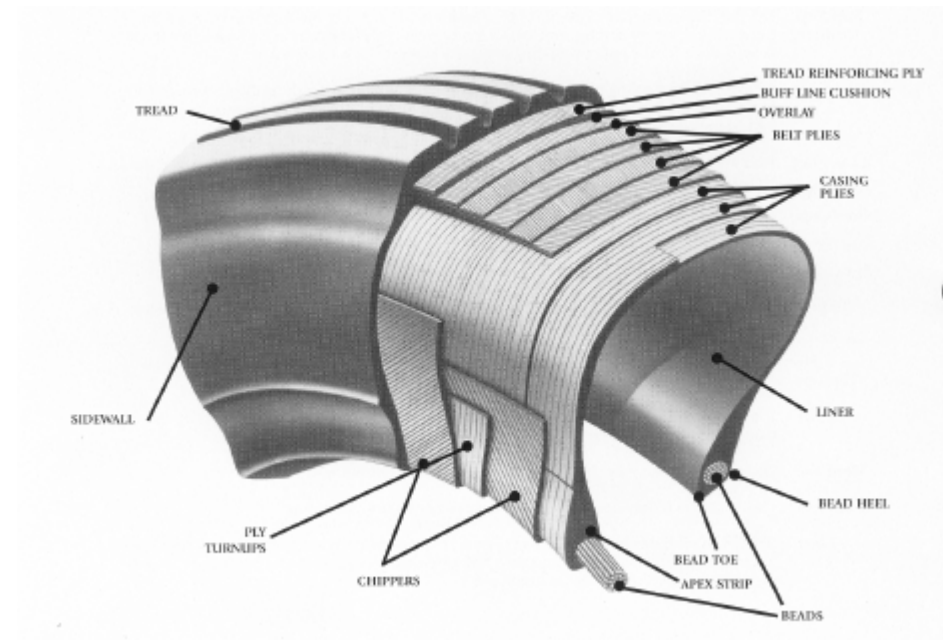
Tyres, Wheels & Brakes



Tyres, Wheels & Brakes



Bias Tyre



Radial Tyre

Tyres, Wheels & Brakes

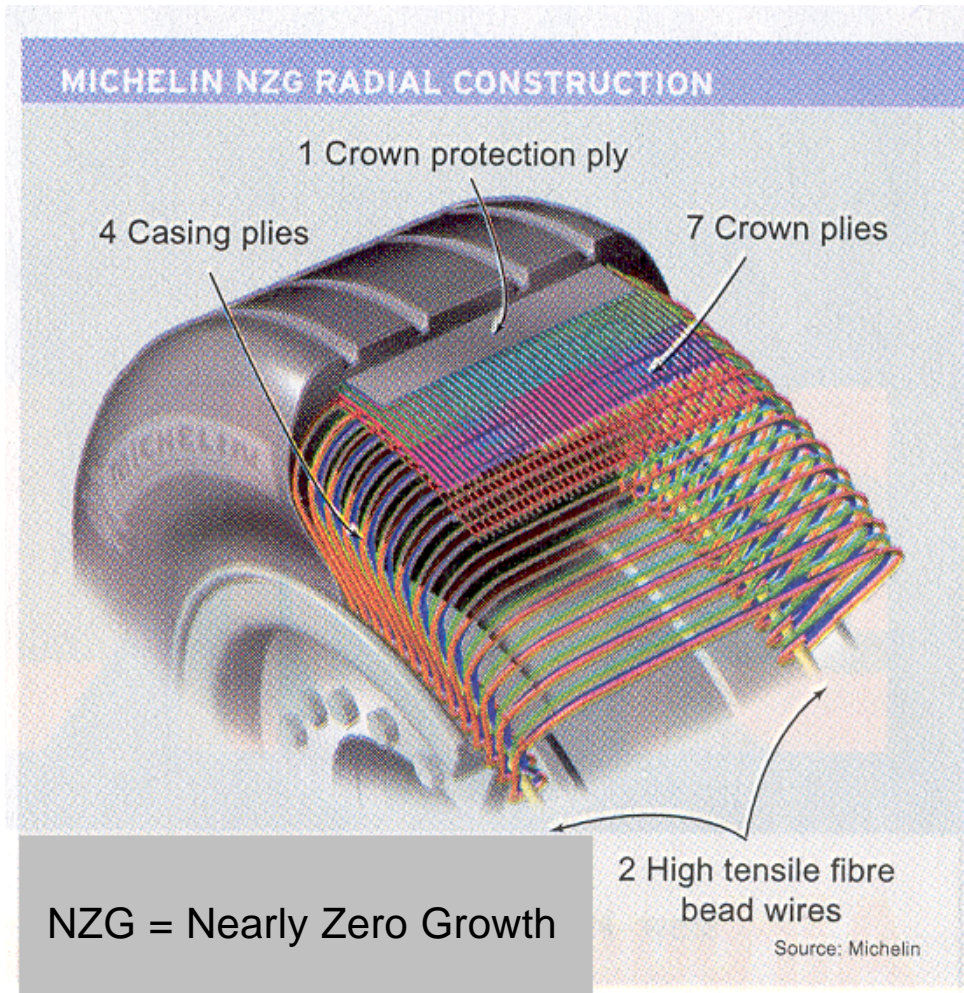


TABLE 1- TYRE CONSTRUCTION DESIGN COMPARISON

	Bias	Radial	NZG Radial
Bead wires	6	2	2
Casing plies	18	7	4
Crown plies	0	9	7
Crown protection plies	2	1	1
Tread grooves	5-6	4	4

Source: Michelin

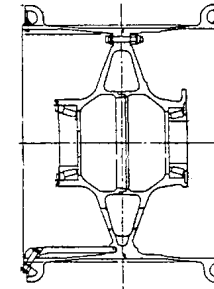
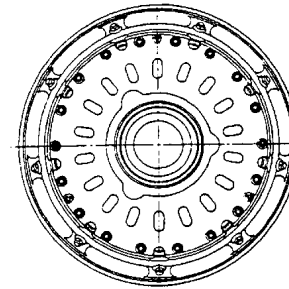
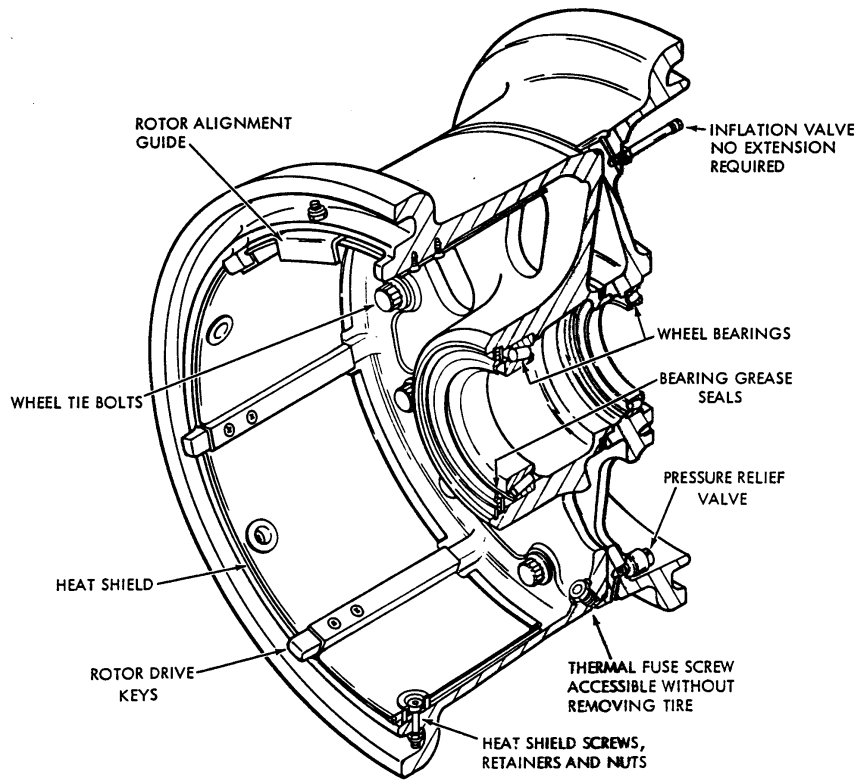
TABLE 2 - TYRE COMPARISON FROM 0-100M/SEC (0-195KT)

Tyre type	Radius increase		
	Centrifugal force	Inflation	Total
Bias	+4%	+8%	+12%
Radial	+2%	+6%	+8%
NZG Radial	+1%	+2%	+3%

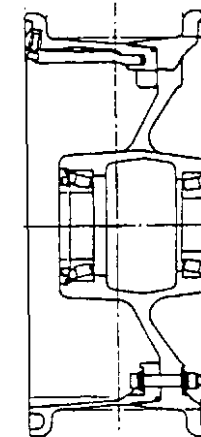
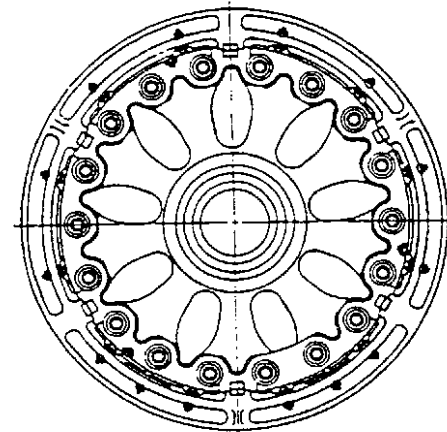
Source: Michelin

NZG Tyre

Tyres, Wheels & Brakes



NLG



MLG

Wheels

Tyres, Wheels & Brakes


 Europäisches Patentamt
 European Patent Office
 Office européen des brevets

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(30) Priority: 15.11.1995 GB 9529446

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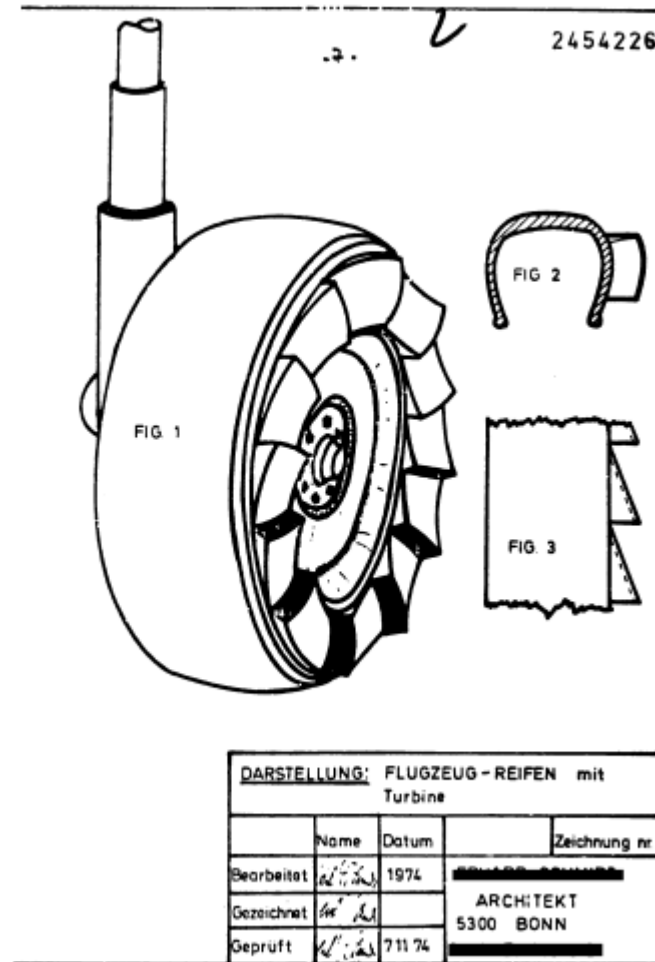
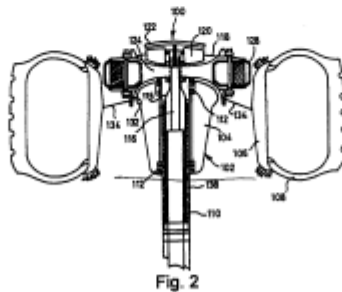
(71) Applicant: Al-Thani, Hamad Ali Jassim Doha, Qatar (QA)

Remarks:
This application was filed on 30/03/99 as a divisional application to the application mentioned under INID code 62.

(54) Apparatus for causing an aircraft wheel to rotate

(57) The invention provides apparatus for causing an aircraft wheel to rotate, comprising at least one nozzle or vent (126, 156, 164, 178, 184) centred on the axle of the wheel and arranged to expel air at least partially circumferentially with respect to said circle, and air supply means for providing air to the or each nozzle or vent (126, 156, 164, 178, 184) for expulsion therefrom, the air supply means comprising a compressed air

source and an air supply passage connecting the compressed air source to the or each nozzle or vent (126, 156, 164, 178, 184), wherein the or each nozzle or vent (126, 156, 164, 178, 184) is rotatable with the wheel such that expulsion of the air through the or each nozzle or vent (126, 156, 164, 178, 184) causes rotation of the wheel.

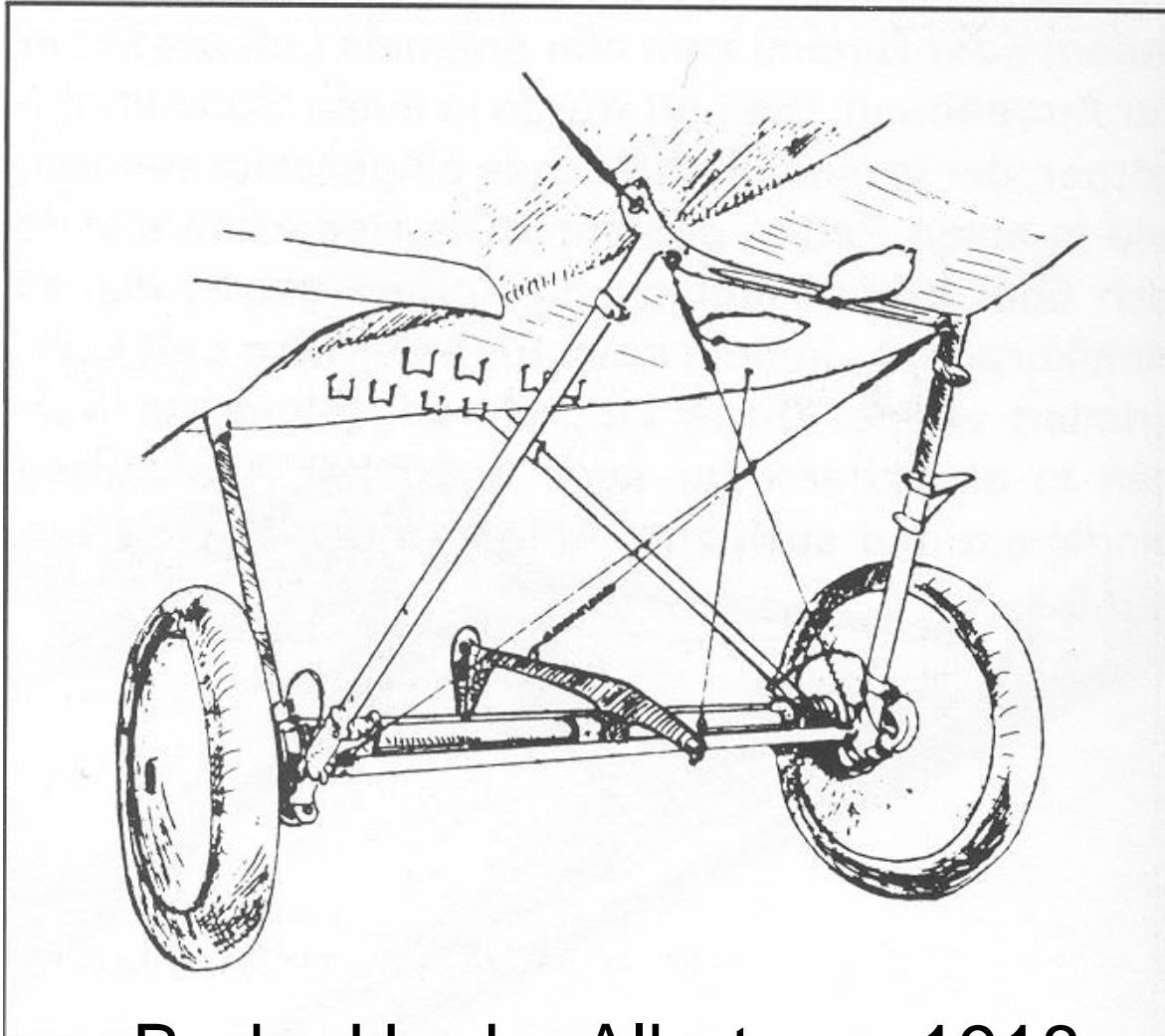


EP 0 934 875 A2

Proprietary Data (P) Business Services 2.16.75.8

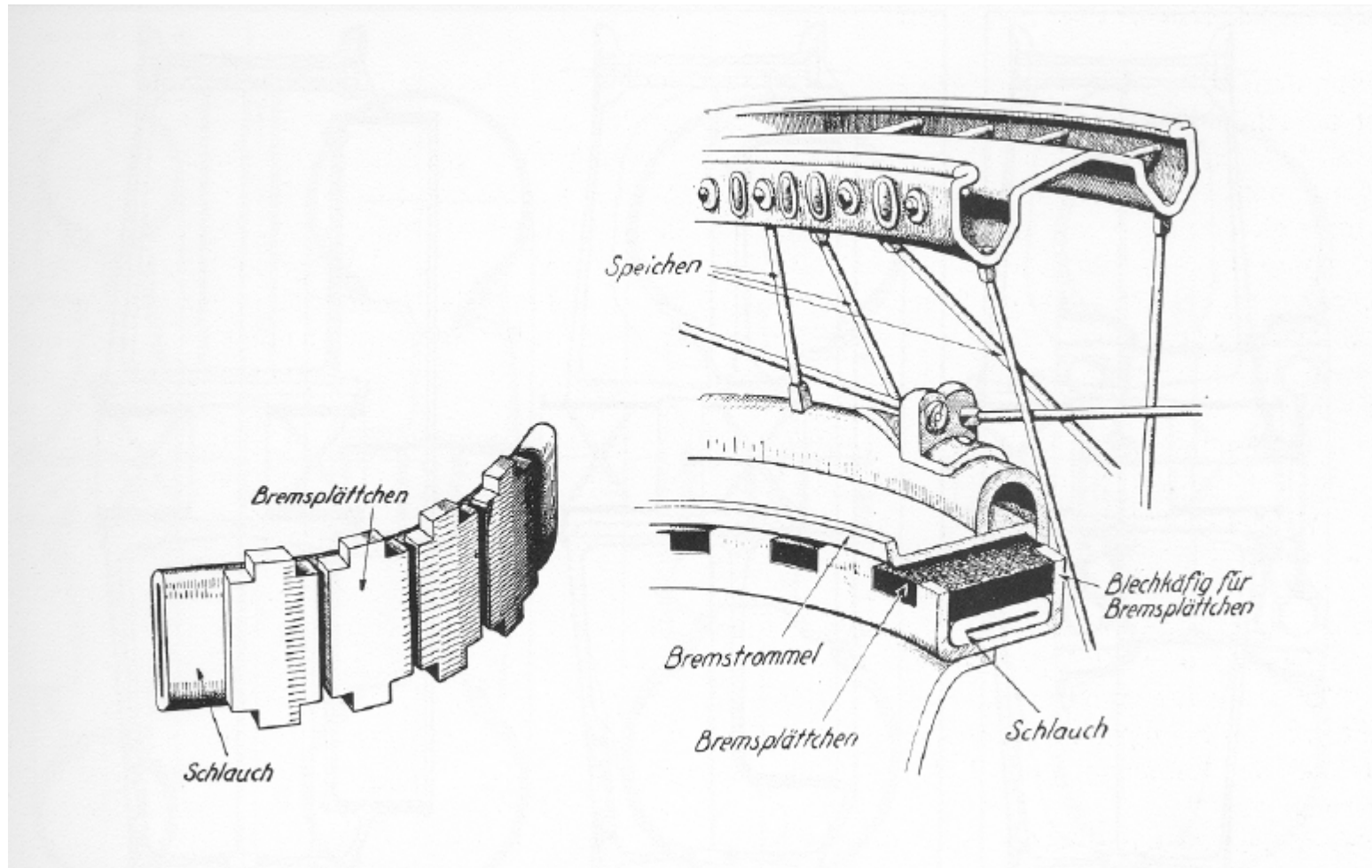
Pre-rotated Wheels

Tyres, Wheels & Brakes



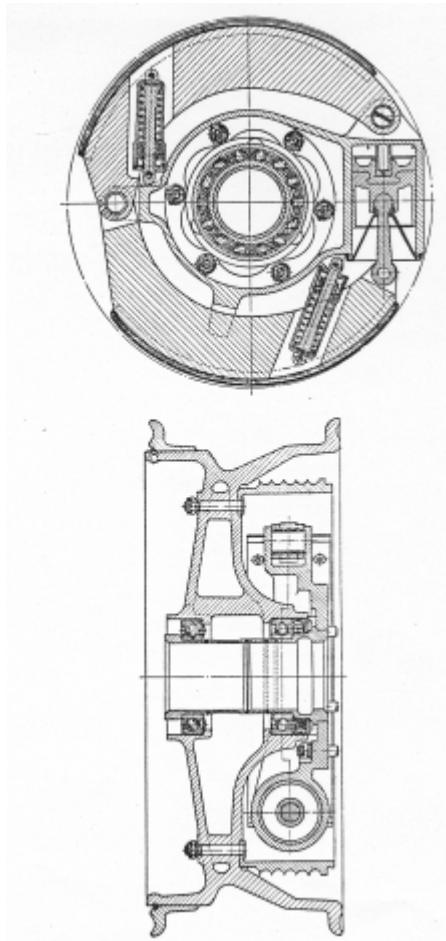
Brake Hook , Albatros , 1918

— Tyres, Wheels & Brakes

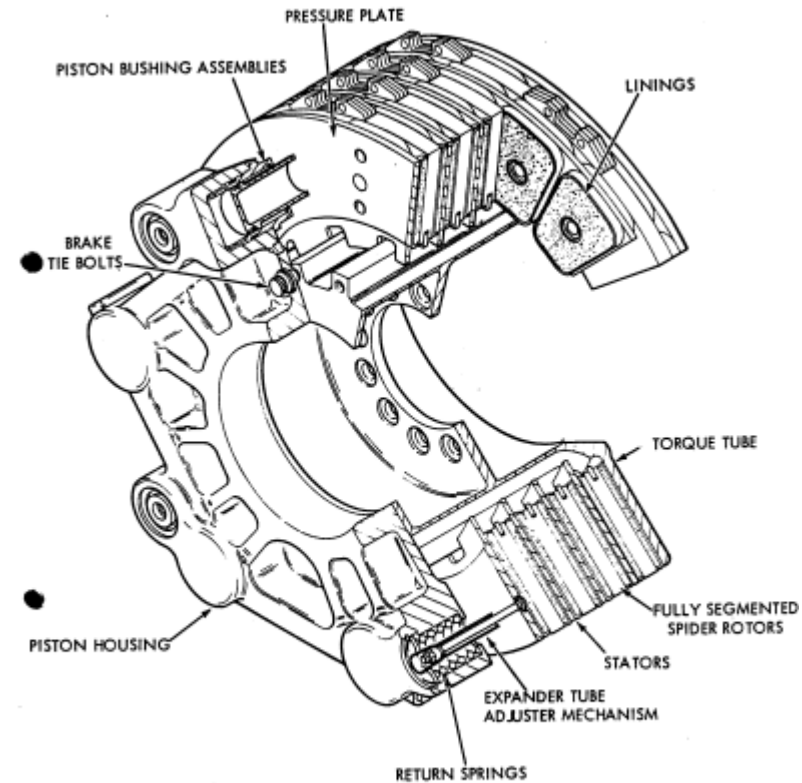


Palmer - Brake Tube

Tyres, Wheels & Brakes

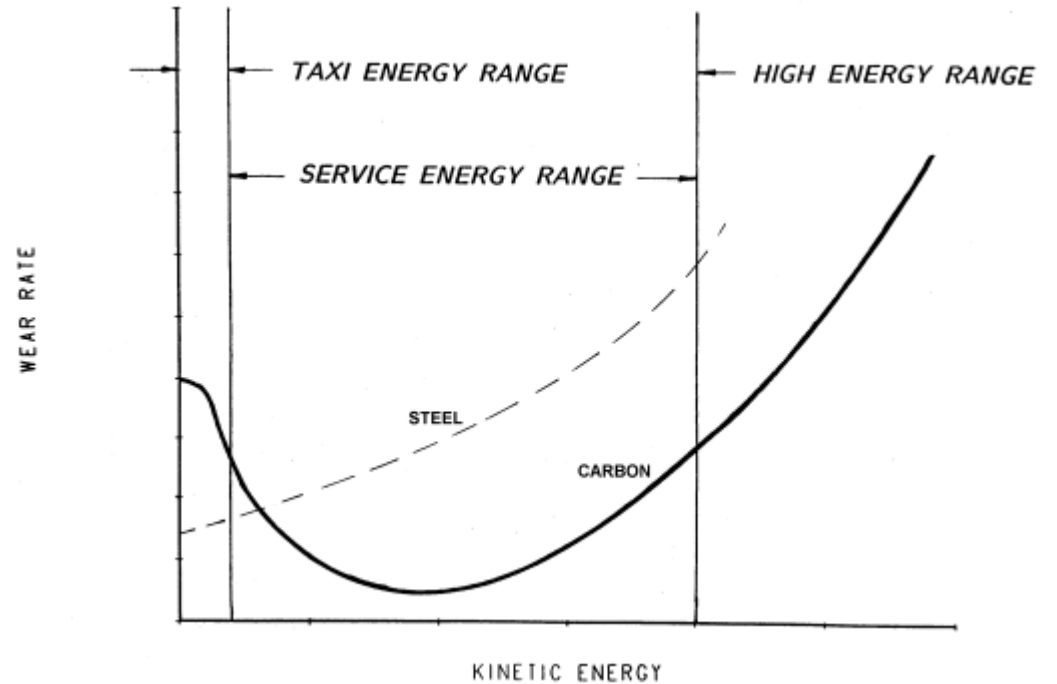
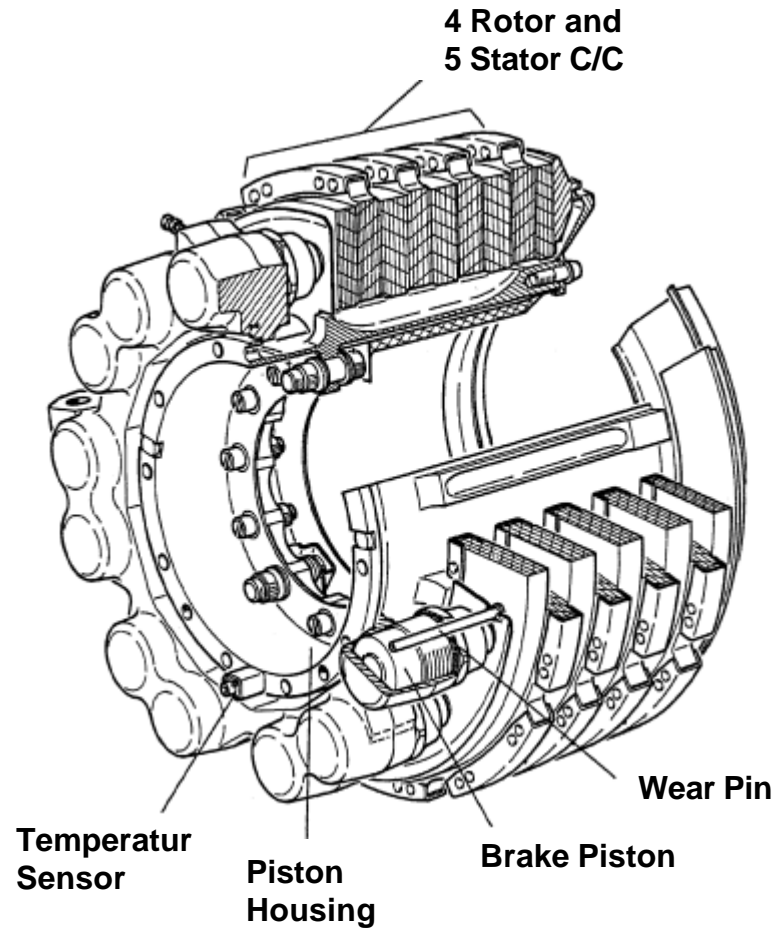


Drum Brake



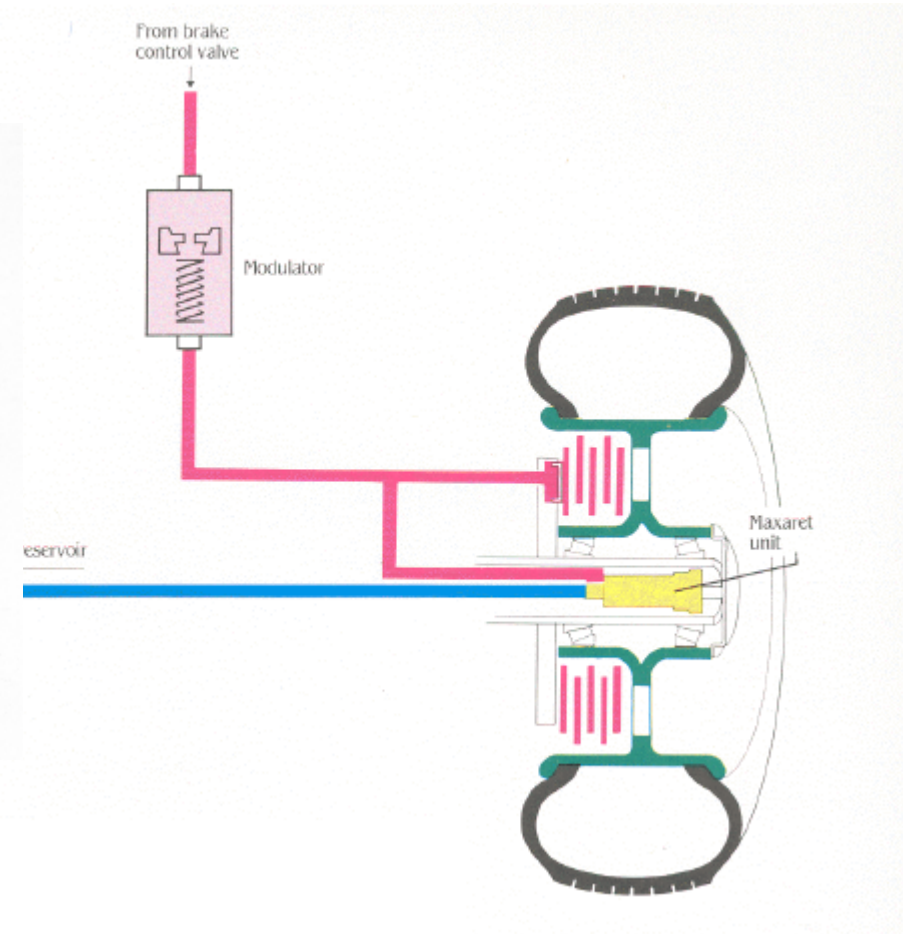
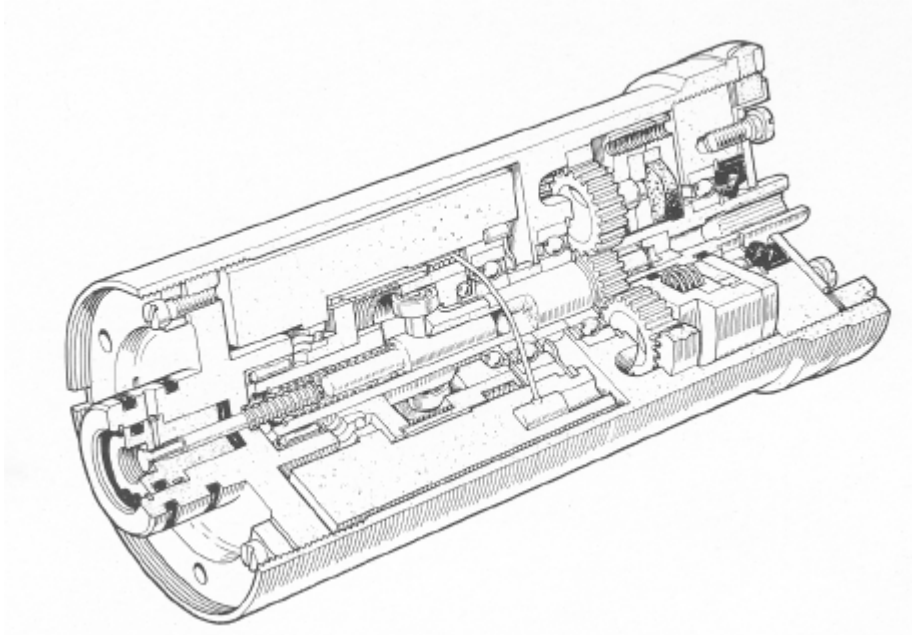
Multi-disc Steel Brake

Tyres, Wheels & Brakes



Carbon Brake

Braking system



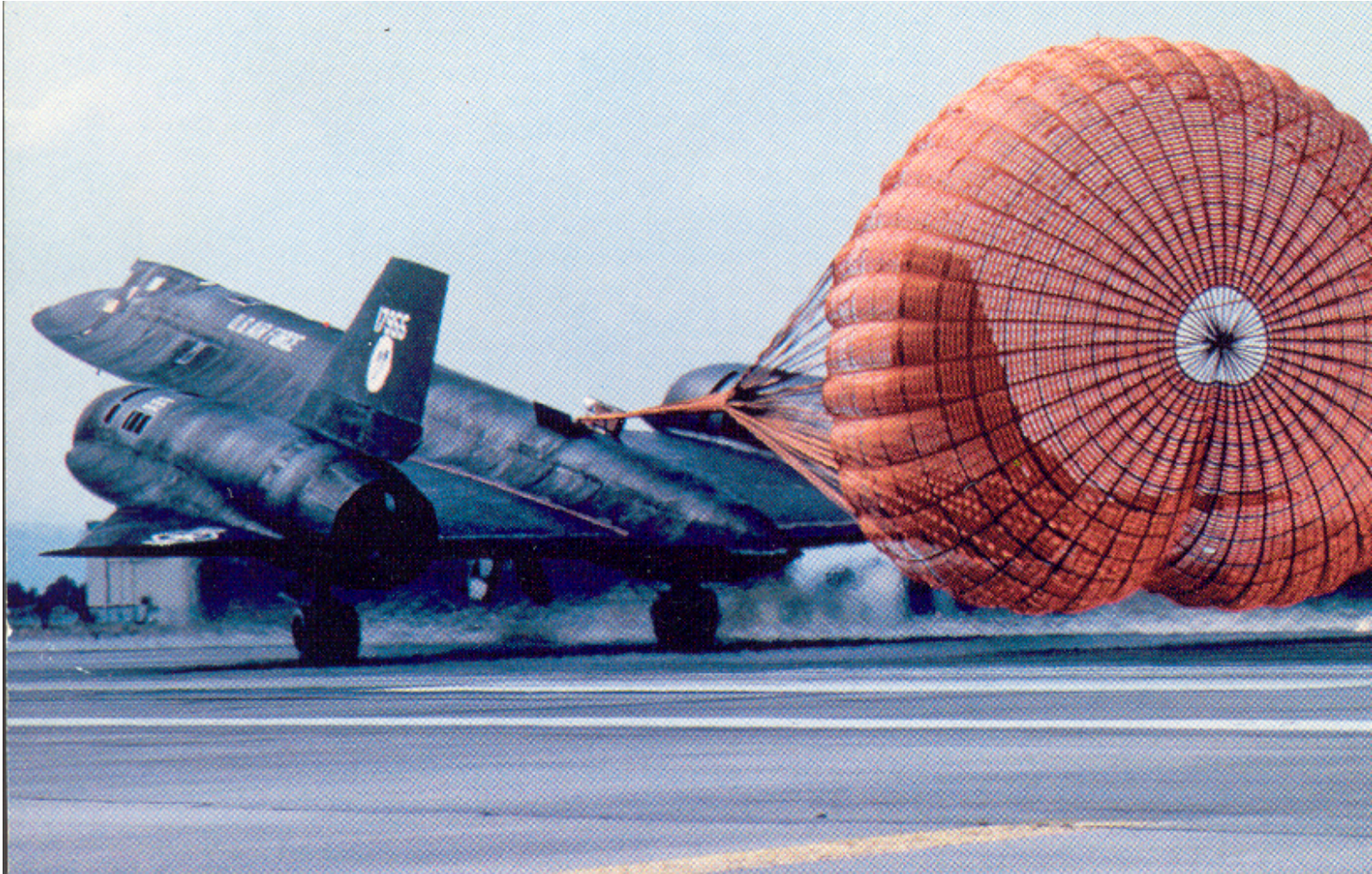
Dunlop Maxaret , 1948

Braking system



Brake Failure

Braking system



Braking Parachute , Lockheed SR-71

Braking system



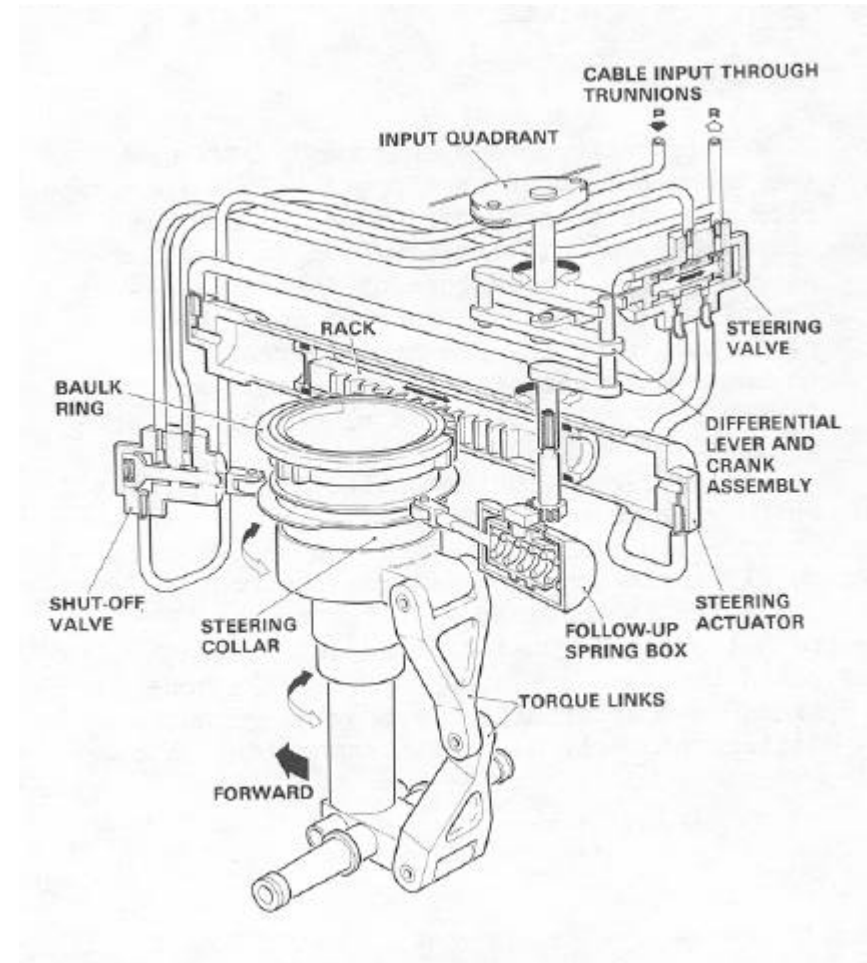
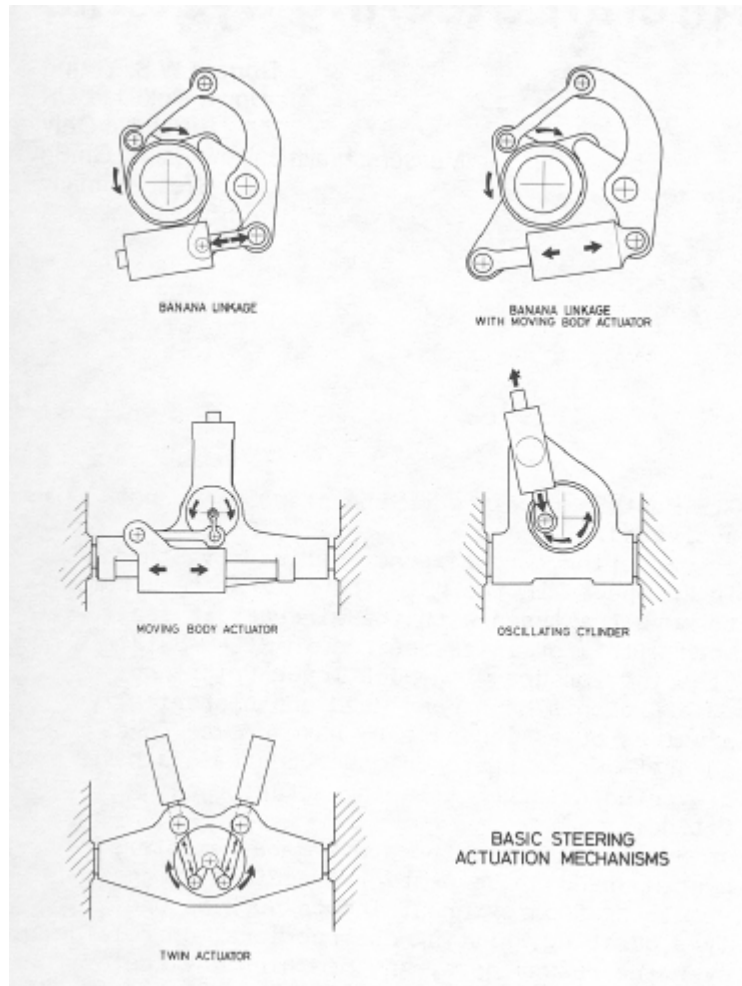
Aircraft Carrier Operations

Braking system



Aircraft Carrier Operations

— Steering system



Steering Mechanism

Actuation system

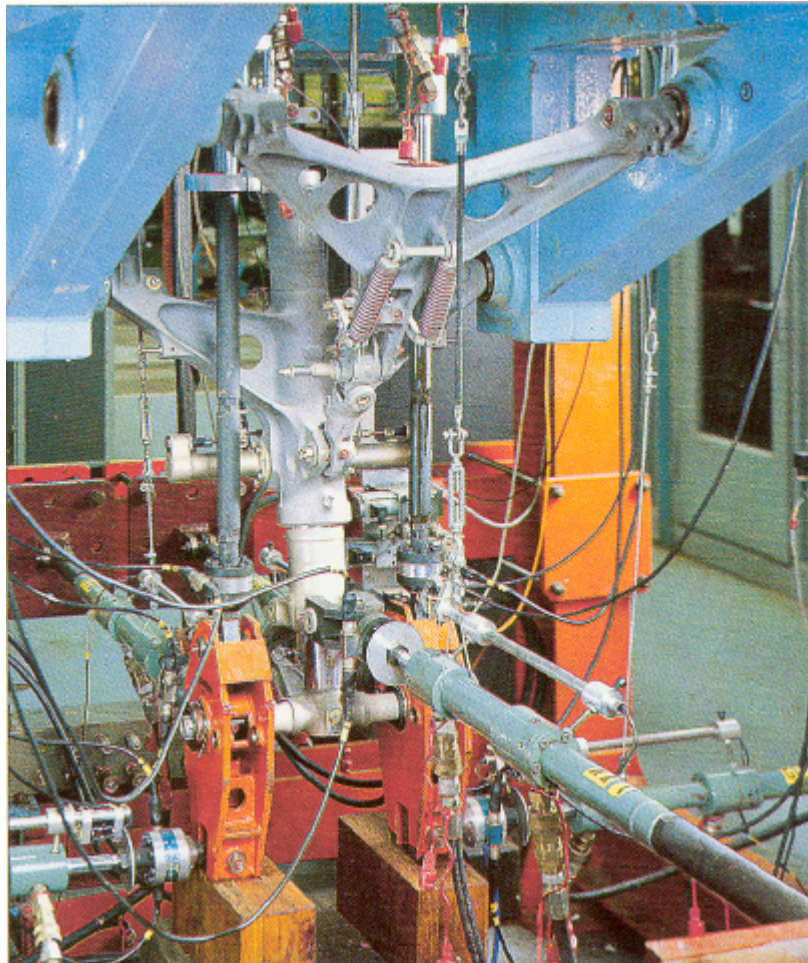


Landing Gear Retraction, Boeing 747

Tests and Certification

- **Landing Gear Certification is part of the Aircraft Certification**
- **Test are performed at the Suppliers facilities**
- **Normally 4 Prototypes are simultaneously used for:**
 - 1. Performance Test**
 - 2. Fatigue Test**
 - 3. Strength Test**
 - 4. Drop Test**

— Tests and Certification

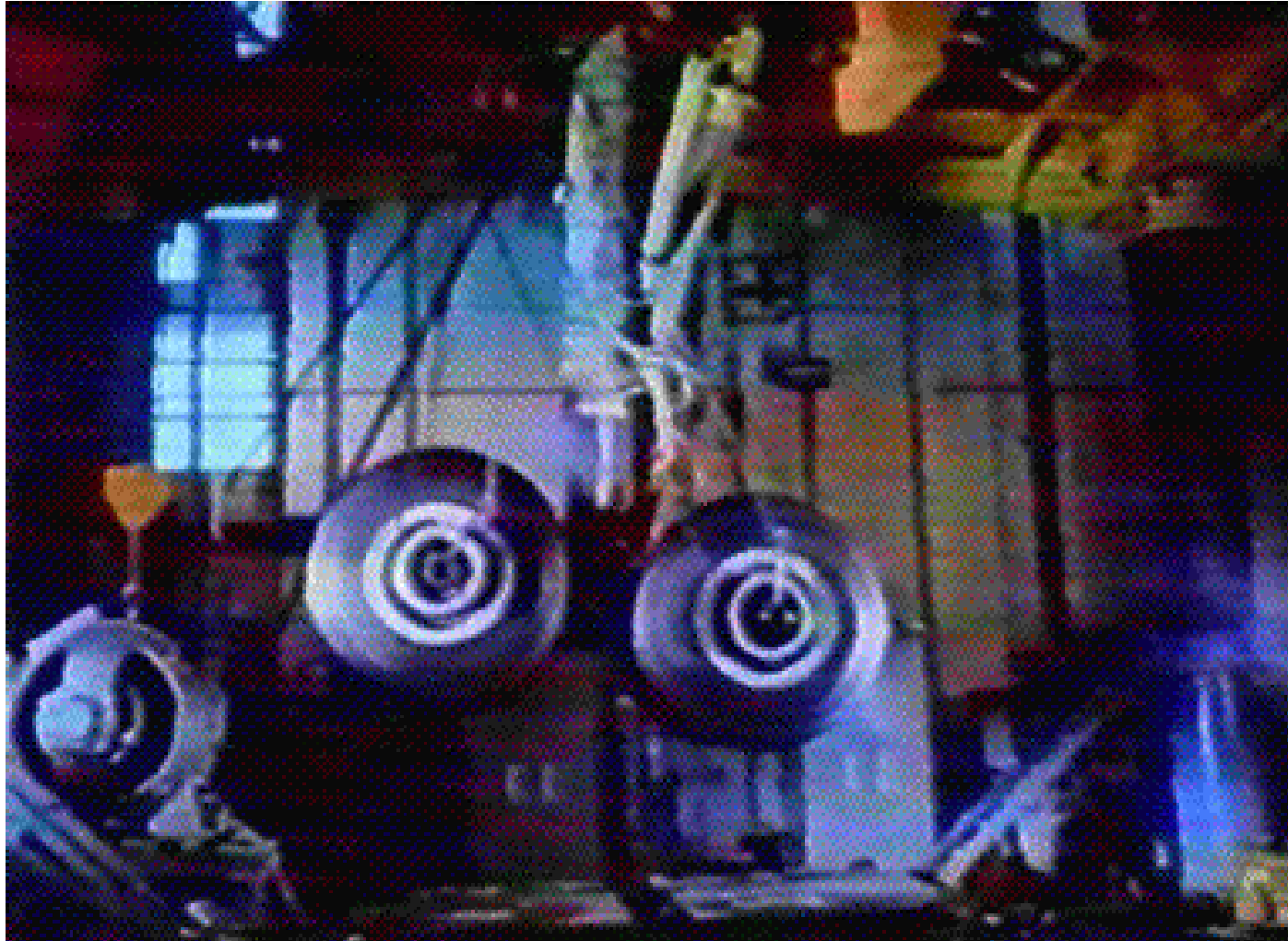


Fatigue Test



Strength Test

Tests and Certification



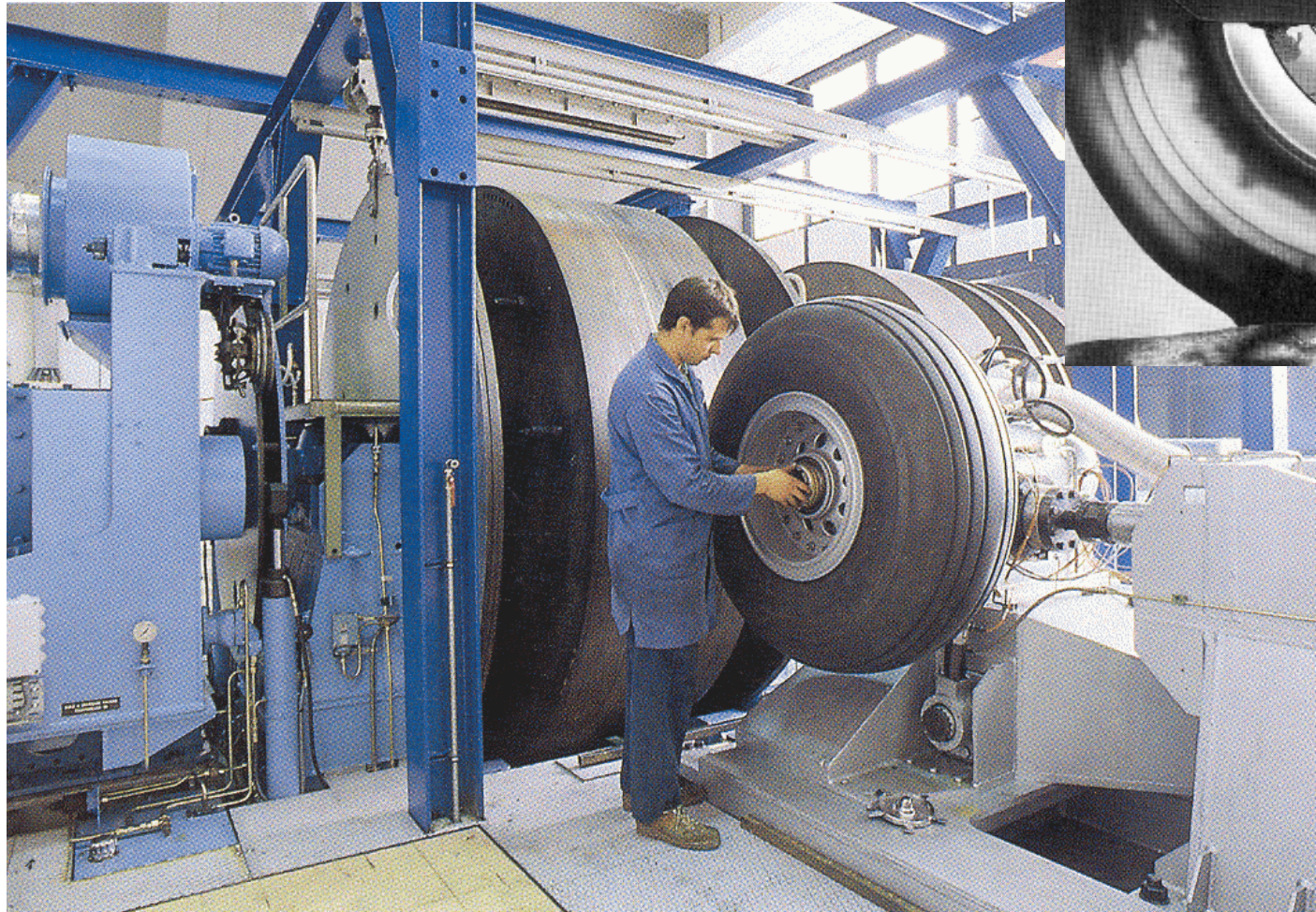
Drop Test , MLG , A310

— *Tests and Certification*



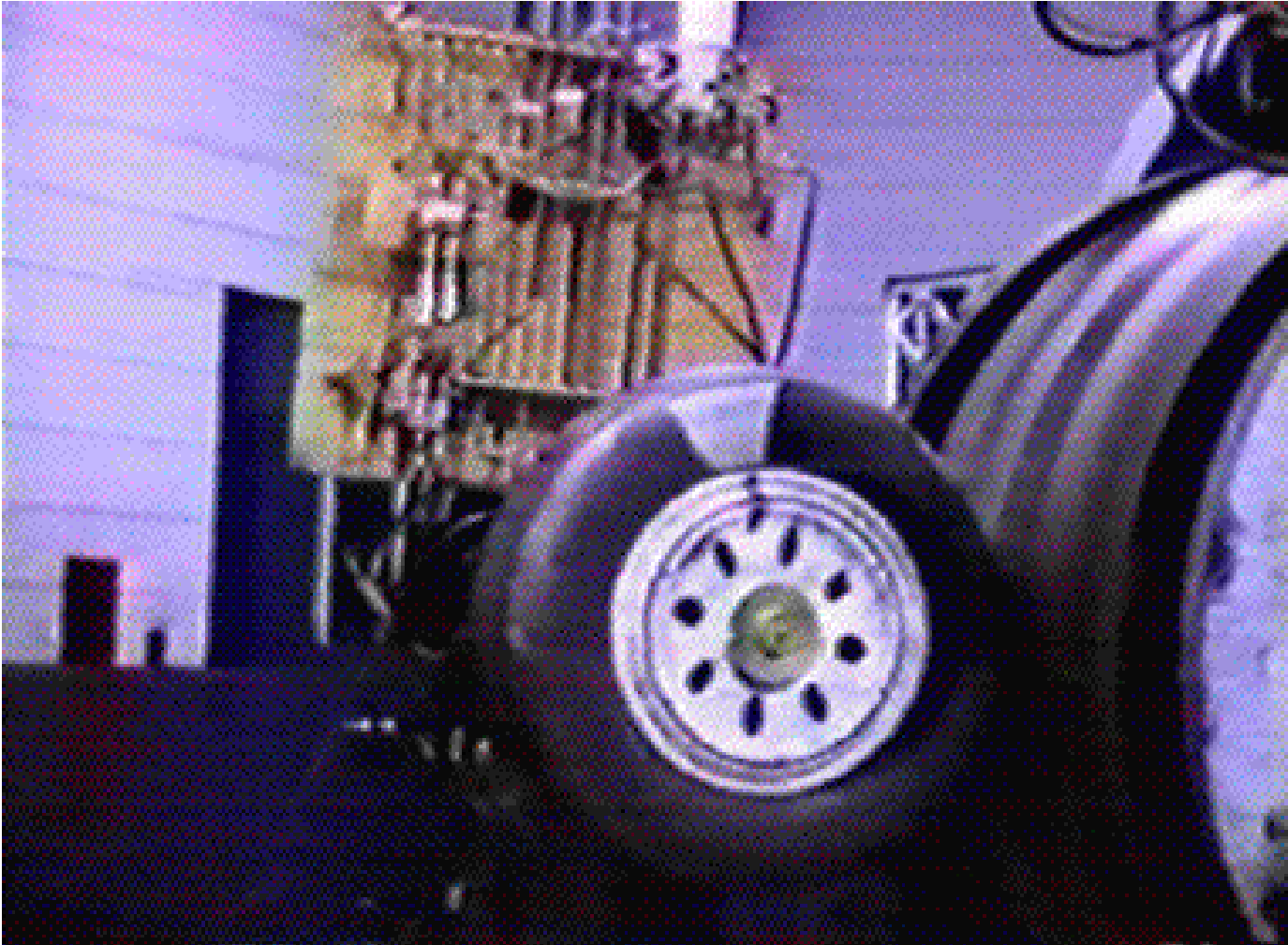
Rejected Take-off , B777

Tests and Certification



Dynamometer

Tests and Certification



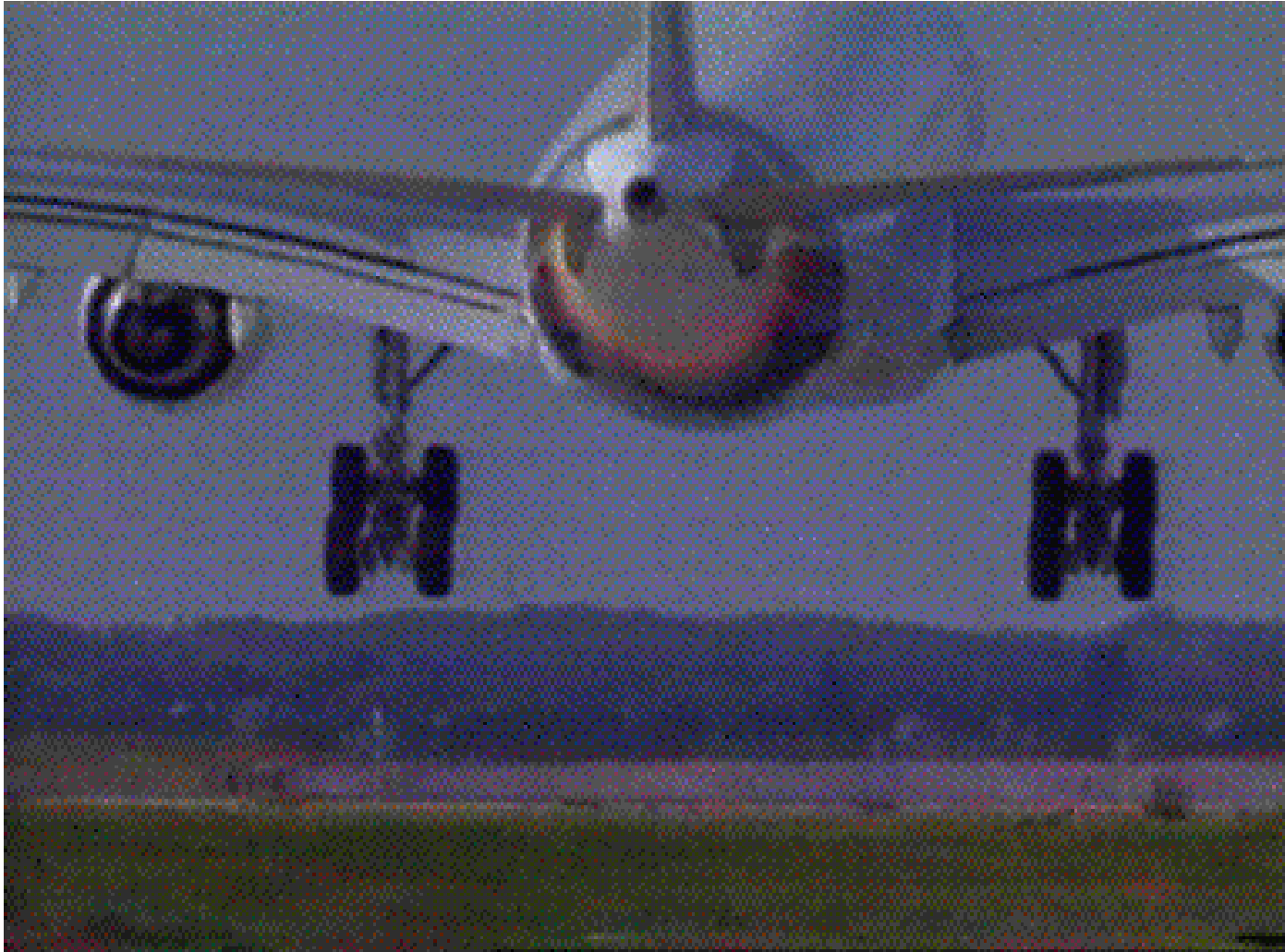
Over-load Tyre Test

Tests and Certification



Tyre Failure

Tests and Certification



Normal Landing , A330

Tests and Certification



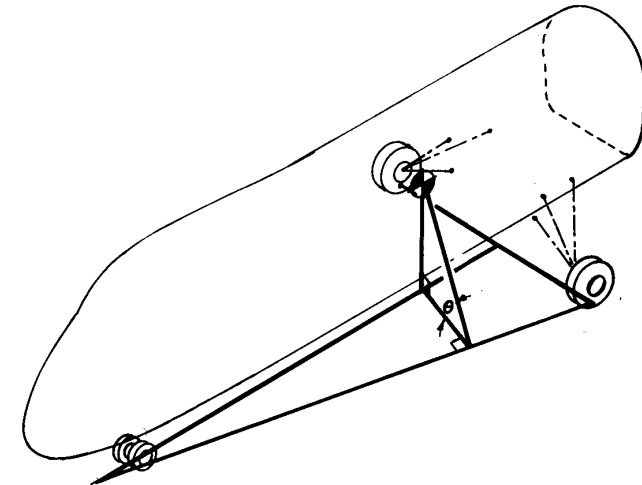
Landing under adverse Conditions , B747-400

— *Tests and Certification*

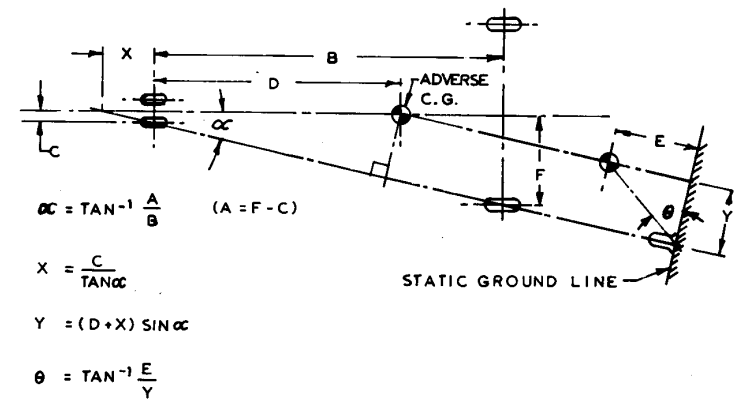
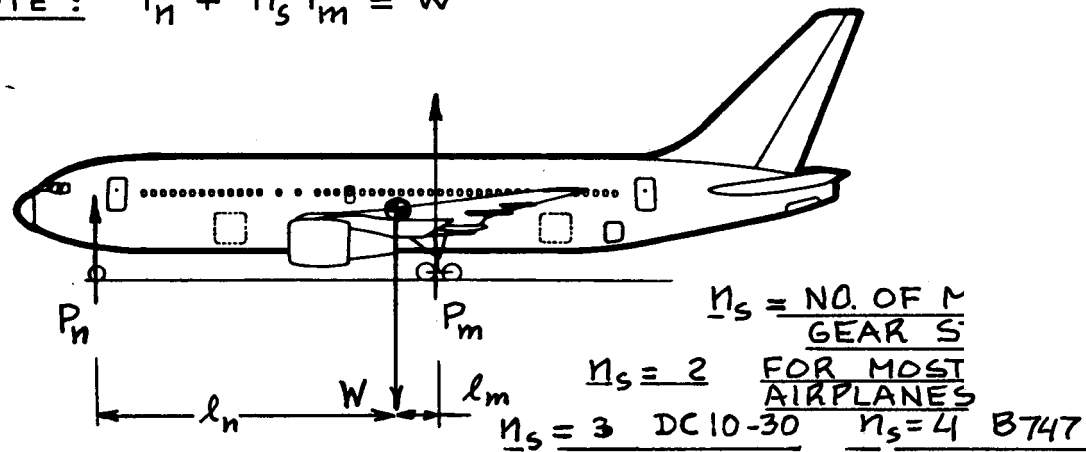


One Engine out Take-off, A340

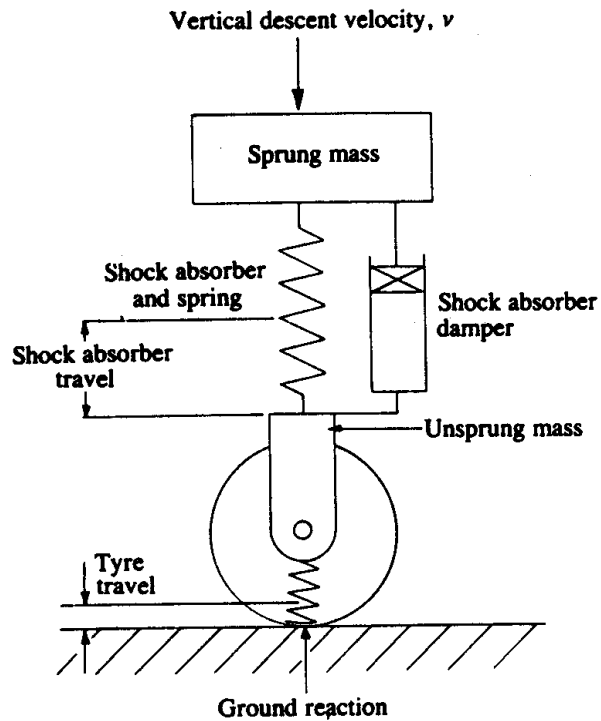
Landing Gear Design



NOTE : $P_n + n_s P_m = W$

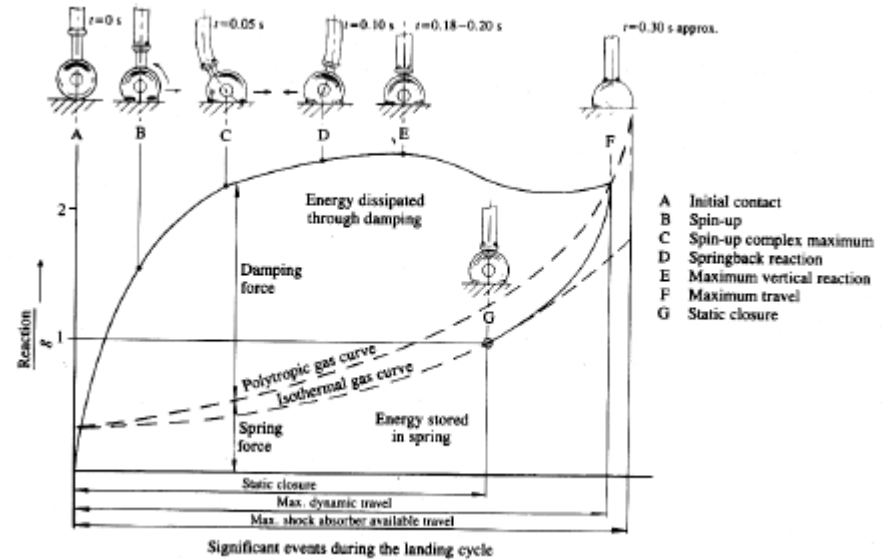


Landing Gear Design



$$\text{Reaction factor, } \lambda = \frac{\text{Ground reaction}}{\text{Associated mass} \times g}$$

$$\begin{aligned} \text{Energy} &= \frac{1}{2} mv^2 \\ &= \lambda \cdot m \cdot g [\text{shock absorber travel} \times \\ &\quad \text{shock absorber efficiency} + \\ &\quad \text{tyre travel} \times \text{tyre efficiency}] \end{aligned}$$



$$(S_t \times n_t \times NW) + (S \times n_s \times NW) = WV^2/2g + (W - L)(S + S_t)$$

tire energy
strut energy
kinetic energy
potential energy

where

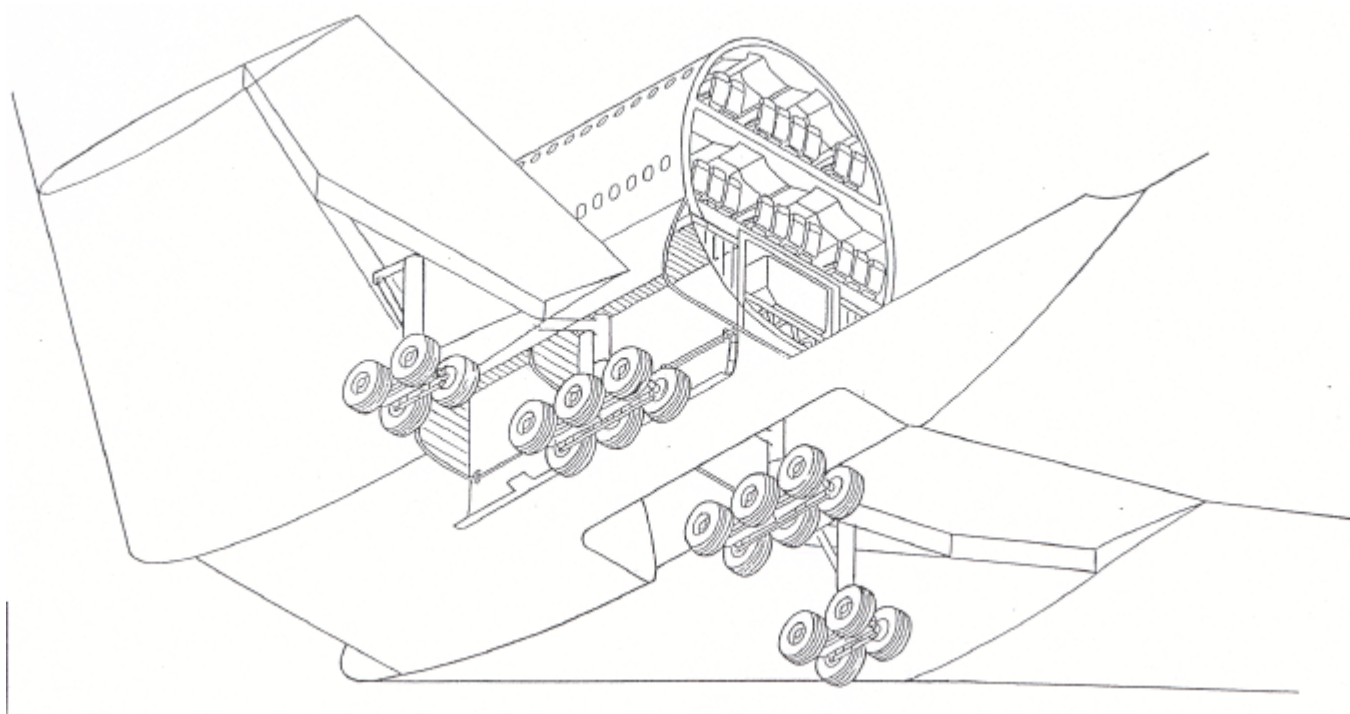
- S_t = tire deflection under N times static load, ft
- S = vertical wheel travel, ft (unknown)
- n_t = tire efficiency, generally assumed to be 0.47
- n_s = shock strut efficiency (assumed initially as 0.80 on an oleo-pneumatic strut)
- N = reaction factor
- W = aircraft weight, lb
- L = lift, lb
- V = sink speed, ft/sec

Landing Gear Design



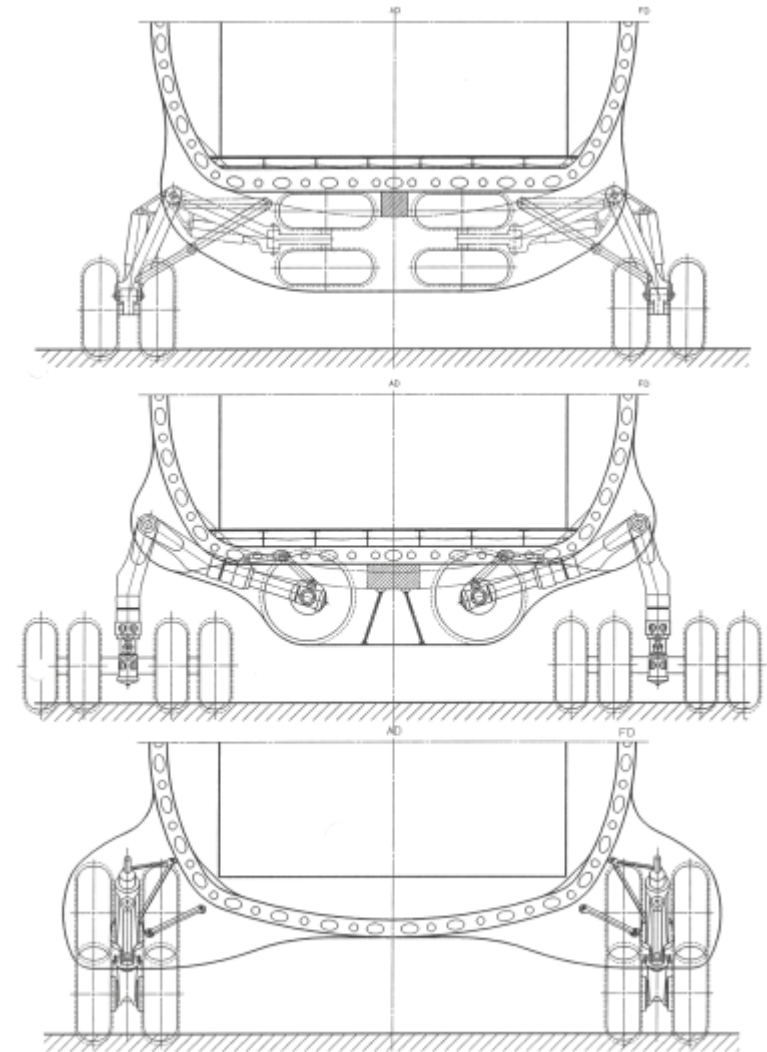
Airbus A380-800

Landing Gear Design



Airbus A380-800

Landing Gear Design



Airbus A400M

Future Trends

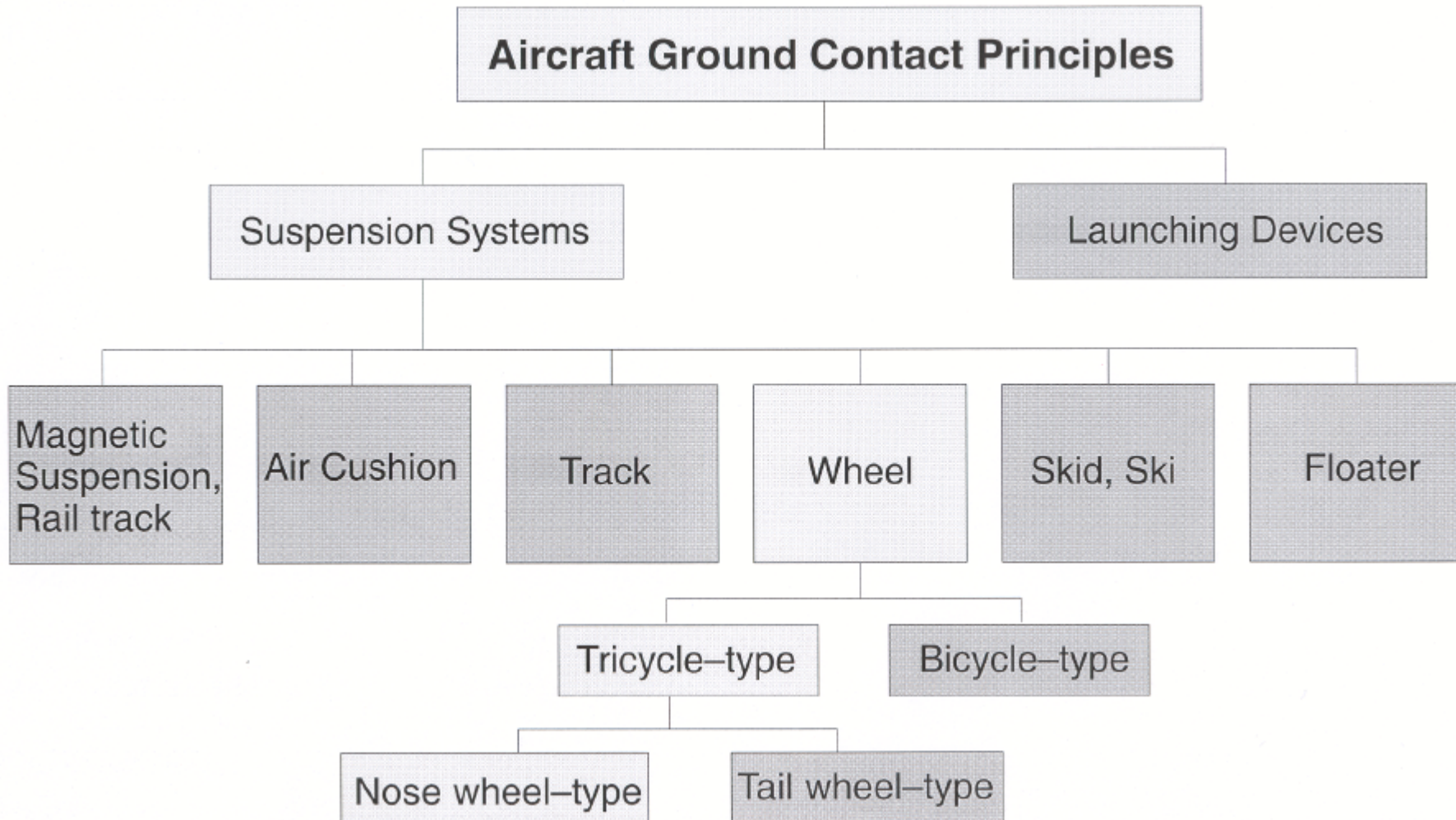
- **Smart Shock Absorbers**
- **Integrated Control Units**
- **Improved Materials for Struts and Brakes**
- **Maintenance-free components**
- **More electrical Actuation**

Special Purpose Undercarriage

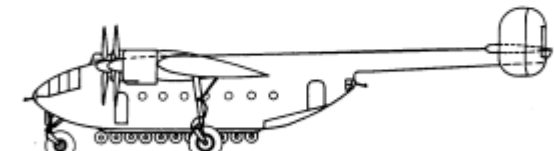
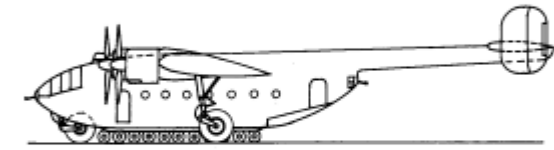
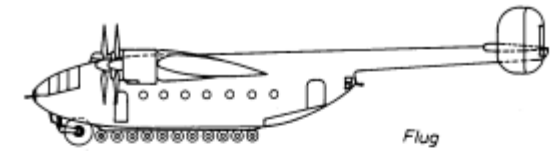


Otto Lilienthal , 1895

— *Special Purpose Undercarriage*

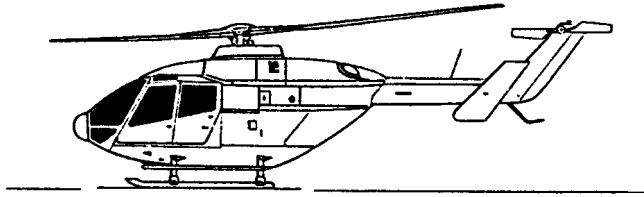


— *Special Purpose Undercarriage*

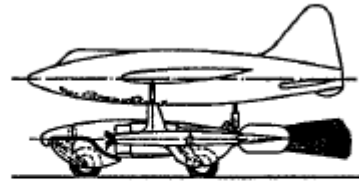


Multi-wheel arrangement , Ar 232

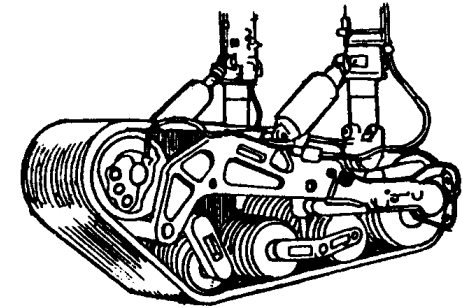
— *Special Purpose Undercarriage*



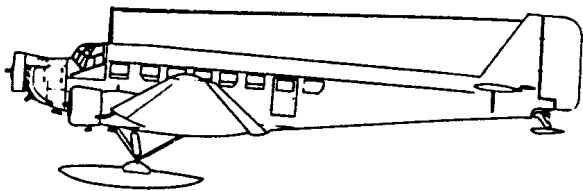
Skid



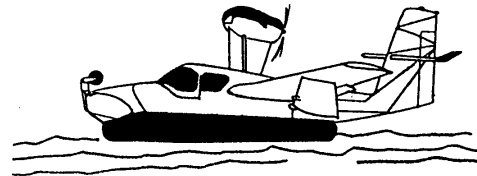
Take-off Trolley



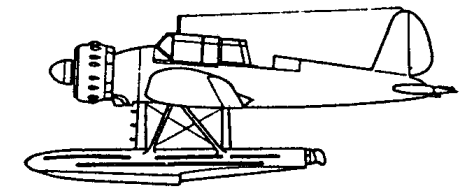
Track



Ski



Air Cushion



Floater

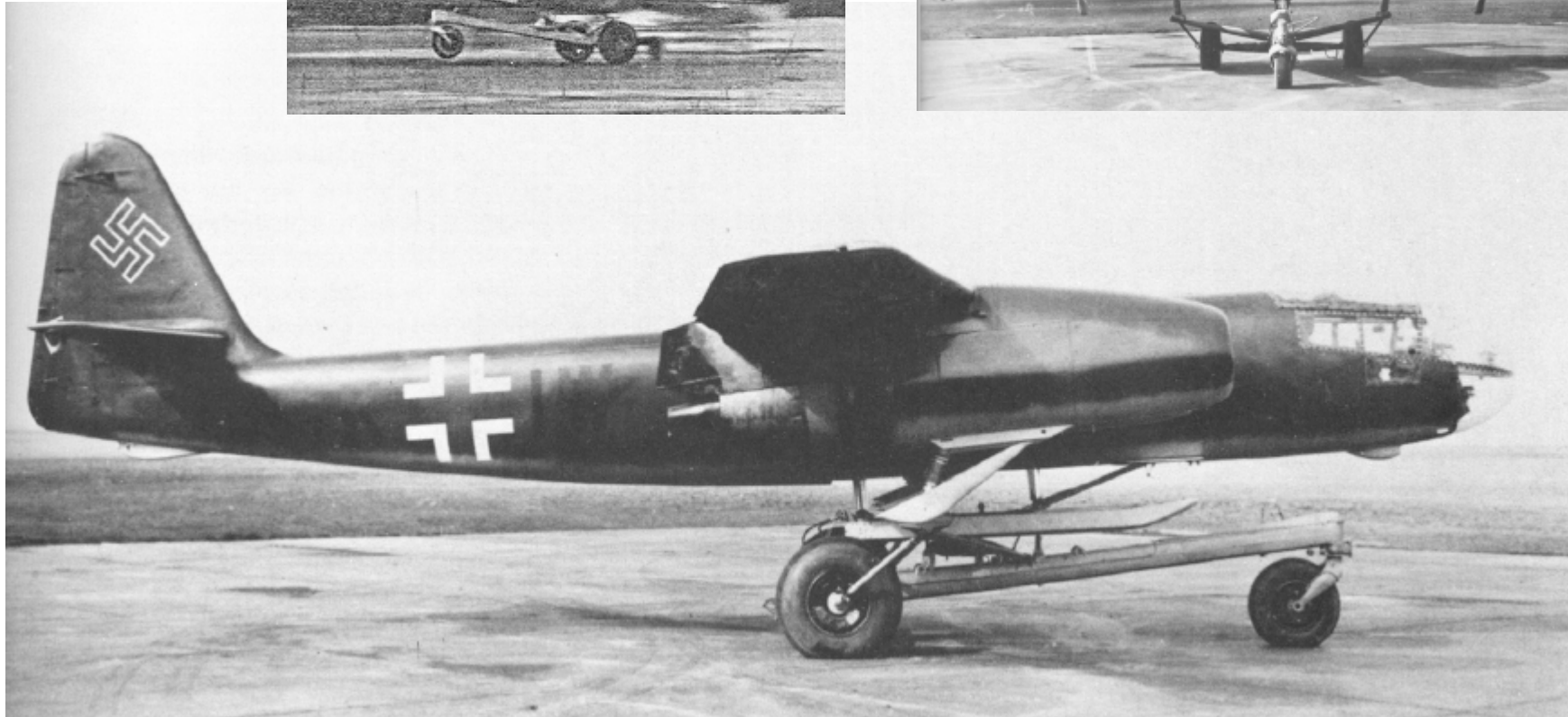
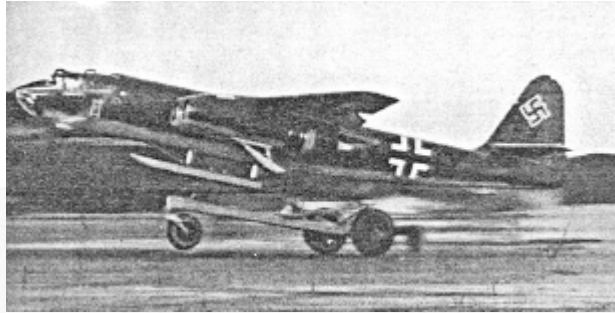
Aircraft Ground Contact Devices

Special Purpose Undercarrige



Skid , EC-135

Special Purpose Undercarriage



Take-off trolley , Ar 234

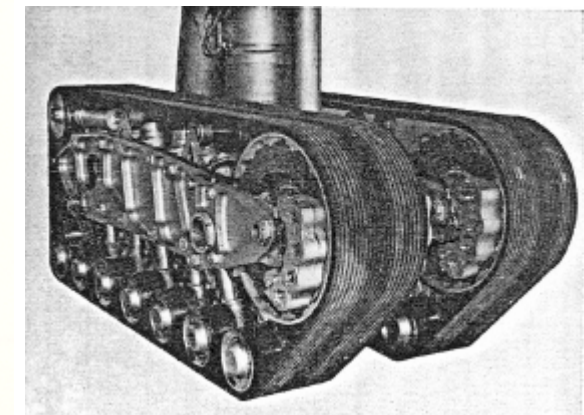
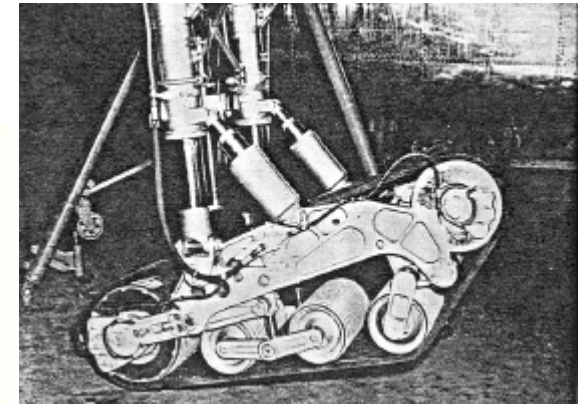
Special Purpose Undercarriage

Bearbeiter: *Handwritten*
 Telefon: 745
 M: 45

Messerschmitt
 AG
 Augsburg

Belastung: $G = 2000 \text{ kg}$
 Geschwindigkeit: $V = 90 \text{ km/h}$
 Bandbreite: ca. $200 \div 220 \text{ mm}$

Federbein
 160
 1150
 Raupenkufe



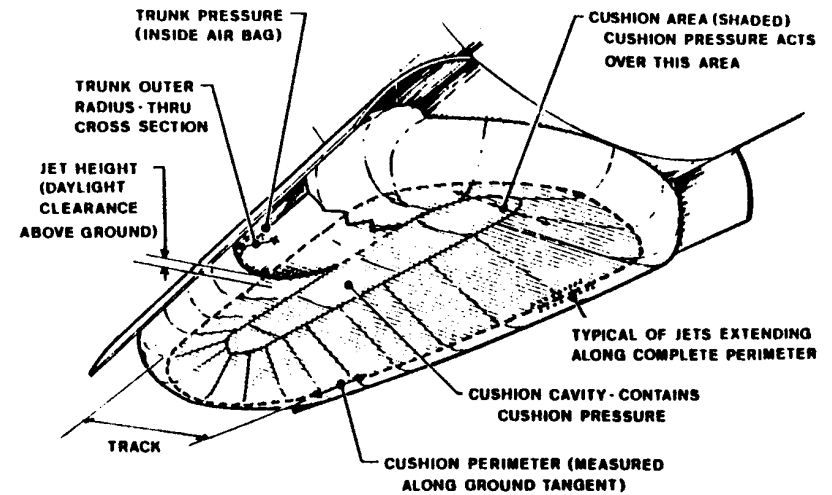
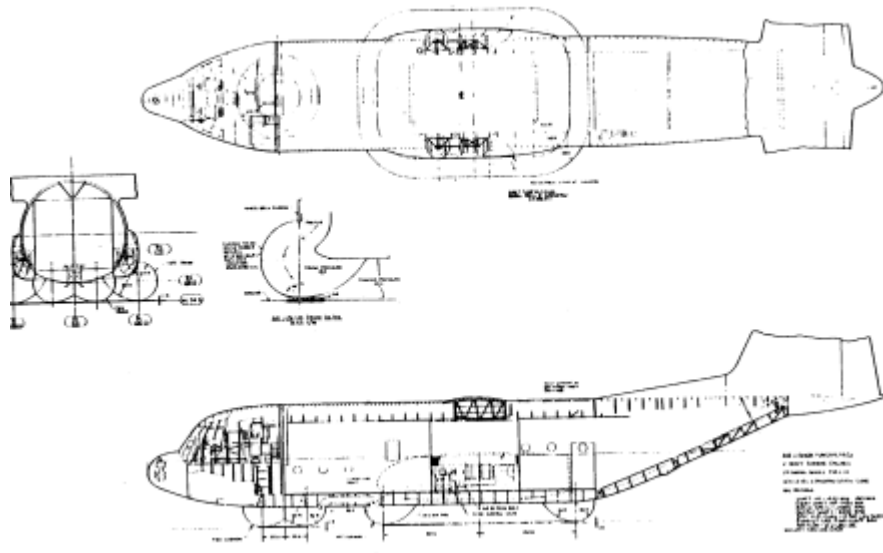
Tracked Gear

Special Purpose Undercarriage



Skies

— *Special Purpose Undercarriage*



Aircraft Ground Contact Devices

Special Purpose Undercarriage



Ekranoplan

Special Purpose Undercarriage



Seaplane - Amphibian

The End

